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RADIO DIRECTION FINDING FIX SOLUTION
IN SPHERICAL TRIGONOMETRY WITH CONFIDENCE
REGION AND LOCATION ERROR ANALYSIS

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Monterey, California



THESIS

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IN SPHERICAL TRIGONOMETRY WITH CONFIDENCE
REGION AND LOCATION ERROR ANALYSIS

by

Ronald Lee Potts

Thesis Advisor:

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March 1973

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Approved for public release; distribution unlimited.



Radio Direction Finding Fix Solution in Spherical
Trigonometry with Confidence Region and Location Error Analysis

by

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Lieutenant, United States Navy
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Submitted in partial fulfillment of the
requirements for the degree of

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March 1973



ABSTRACT

Although parts of this paper may be applicable to various other location finding systems, radio direction finding was selected for development because of the extensive literature available on the subject. Using a modification of the navigation coordinate system, equations in spherical trigonometry are developed by analogy with graphical solution procedures. Ellipticity of the earth is corrected for by use of geocentric latitude in all calculations. The least squares solution is found by a combination of second-order gradient searches and Golden Section Searches on the corrected sphere. A Chi-Square contour confidence region is presented with defense against the empty region arguments and with verification of its reliability of target containment. Sensitivity of location error to network layout, number of measurements available and intercept geometry factors is examined by computer simulation, demonstrating the predictive value of these statistics. It is shown that the correlation between location error and the size of a reliable confidence region is too minimal to have predictive value.

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I. INTRODUCTION

Various procedures have been described for estimating an emitter location from a set of measured signal azimuths reported to a control station by dispersed direction finding (DF) sites. Such procedures range upward in complexity from visual estimates off a plot of great circle arcs on a gnomonic projection of the operating area of potential targets. Other procedures exist for describing confidence criteria about the location estimate. The present paper was motivated by an interest in confidence regions and in location error analysis. Such studies require either that the procedure used be free of approximating assumptions, or that the variability of the approximations be included in the confidence criterion. The former course was preferred for simplicity and consistency of estimates. This paper presents an emitter location estimate procedure based only upon known DF site coordinates and signal azimuths. The procedure described in sections II and III converges quickly to a least squares solution and does not require an initialization point estimate of the location from external sources (Pope, 1971). The procedure does not require a rejection of outliers (McCalla, 1970) in order to converge to the least squares solution. The approach is free from assumptions in the categories that the earth is flat in the solution region, that the earth is a perfect sphere, that great circle arcs

are well enough approximated by straight lines of position, and that azimuth errors are well enough approximated by parallel displacements of such lines of position (Daniels, 1951).

Section IV describes a Chi-Square contour confidence region procedure. Factors from intercept geometry are discussed in section V. Section VI is a description of procedures used to simulate the problem for the purpose of the location error analyses described in sections VII and VIII. An alternative to confidence regions for predicting location accuracy is discussed in the concluding section.

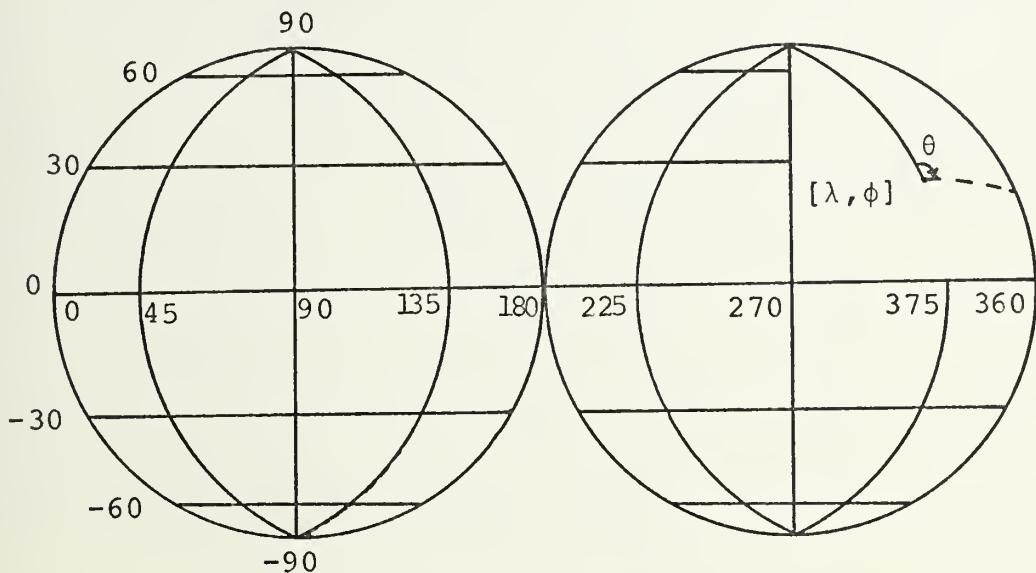
II. MATHEMATICAL DESCRIPTION OF SPATIAL RELATIONSHIPS

Locations and spatial relationships of interest in direction finding and target fixing can be expressed completely in spherical trigonometry. The advantage of such expression is that each mathematical step has its simple graphical counterpart. This paper contains notational and descriptive departures from other direction finding models based on spherical trigonometry. Such notational changes follow those used in geodesy for describing locations and spatial relationships and those used in textbooks on spherical trigonometry for developing equations to express such relationships. Equations are restricted to sine and cosine functions only in order to simplify calculating first and second order derivatives of somewhat lengthy equations.

A. GLOBAL COORDINATE SYSTEM

The coordinate system is a slight modification of the system used in mapping. Longitude, denoted λ , is measured from zero at the prime meridian eastward through 360 degrees. Latitude, denoted ϕ , is measured from zero at the equator northward through ninety degrees and southward through minus ninety degrees. Azimuth, denoted θ , is measured from zero at true north clockwise through 360 degrees. Such a coordinate system is sufficient to represent any point on the globe as well as the great circles which represent ideal wave propagation paths. Mathematical descriptions of spatial

relationships must account for the longitude discontinuity at the prime meridian and for the azimuth discontinuity at true north. A great circle arc with azimuth angle indicated is sketched on the coordinate system in Figure 1.



Global Coordinate System

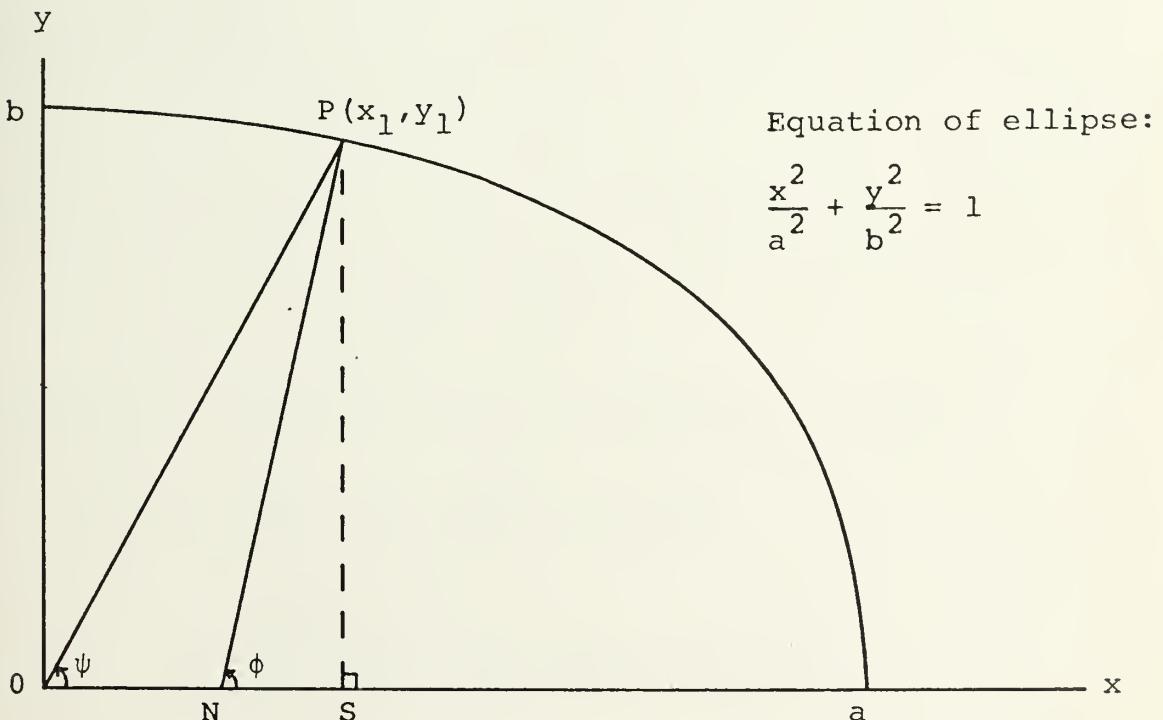
FIGURE 1

B. GEOCENTRIC LATITUDE

Geographical latitude, used in navigation and mapping, is based upon astronomical observations and mathematical deductions that do not correct for the ellipticity of the earth. Consequently, geographical latitude is slightly greater in magnitude than the arc on a meridian from the equator to a location with vertex angle at the center of the earth. Rules of spherical trigonometry can be more accurately applied to describe spatial relationships following the conversion from geographical latitude to geocentric, or earth-

centered, latitude. Assuming that waves propagate along great circle arcs, then no information is lost since such paths are geometrically independent of radii to the earth center from both the signal emitter and from the DF site. The maximum discrepancy between the two types of latitude occurs at ± 45 degrees latitude and is about equal to 11.5 minutes of arc. Although the conversion equation appears in books on geodesy, a derivation could not be found, so the equation was derived for reference purposes.

In Figure 2, a and b represent the semi-major and semi-minor axes of the ellipsoid. The point $P(x_1, y_1)$ represents any point on the ellipsoid. The line OP connects P to the origin at O , the earth center. SP is perpendicular to the



Relationship Between Geographical and Geocentric Latitude

FIGURE 2

equatorial plane, the x-axis. NP is the normal line to the ellipsoid at P, and the angle PNS is equivalent to geographical latitude. Angle PON is equivalent to geocentric latitude, denoted ψ . Ellipticity of the quadrant is exaggerated in order to distinguish the two angles.

From Figure 2, it can be seen that

$$SP = OS \tan \psi, \text{ and that } SP = NS \tan \phi,$$

so that

$$\tan \psi = \frac{NS}{OS} \tan \phi \quad (2-1)$$

From the equation of the ellipse,

$$y_1 = \frac{b}{a} (a^2 - x_1^2)^{\frac{1}{2}} \quad (2-2)$$

The equation of the normal line NP is given by

$$y - y_1 = - \frac{1}{f'(x_1)} (x - x_1) \quad (2-3)$$

Differentiating equation (2-2) to solve $f'(x_1)$,

$$f'(x_1) = - \frac{b}{a} x_1 (a^2 - x_1^2)^{-\frac{1}{2}}$$

Substituting in equation (2-3) and rearranging terms,

$$\begin{aligned} y - y_1 &= \frac{a}{bx_1} (a^2 - x_1^2)^{\frac{1}{2}} (x - x_1) \\ &= \frac{a^2 y_1}{b^2 x_1} (x - x_1), \text{ from equation (2-2).} \end{aligned}$$

Then at $y = 0$, y_1 can be cancelled from both sides, and

$$\frac{x - x_1}{x_1} = - \frac{b^2}{a^2}$$

From Figure 2, $x_1 = OS$, and at $y = 0$, $x = ON$, so that
 $x_1 - x = NS$.

$$\frac{NS}{OS} = \frac{b^2}{a^2}$$

Substituting this last result in equation (2-1),

$$\tan\psi = \frac{b^2}{a^2} \tan\phi \quad (2-4)$$

The conversion is usually stated as a function of earth eccentricity,

$$e = \frac{1}{a} (a^2 - b^2)^{\frac{1}{2}}$$

Then

$$e^2 = \frac{a^2 - b^2}{a^2}$$

$$= 1 - \frac{b^2}{a^2}$$

Therefore,

$$\frac{b^2}{a^2} = 1 - e^2, \text{ and it follows that}$$

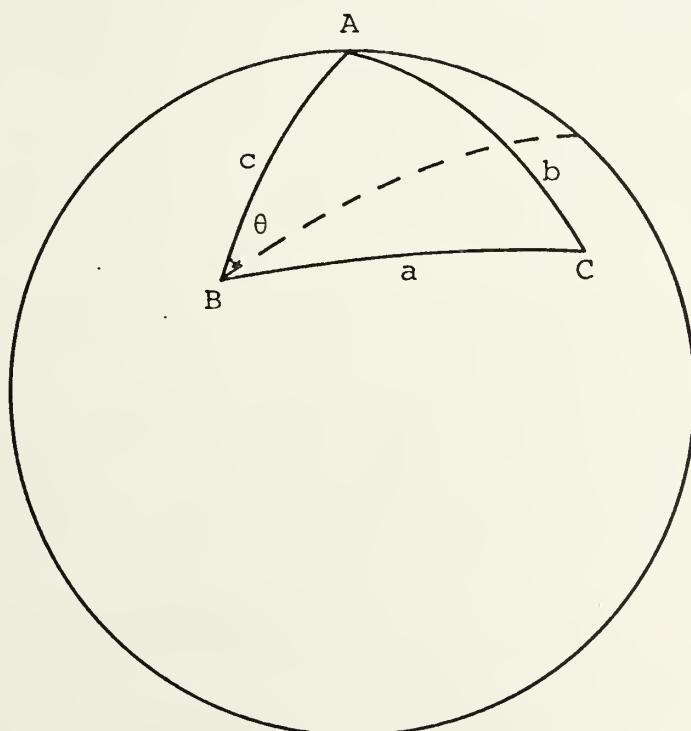
$$\tan\psi = (1 - e^2) \tan\phi \quad (2-5)$$

Using $e = 0.081813$ from the International Ellipsoid of Reference (reference 3),

$$\tan\psi = 0.99327733 \tan\phi \quad (2-6)$$

C. THE POLAR TRIANGLE

The mathematical description of spatial relationships is based upon a polar triangle constructed from the great circle arcs (measured by the angle subtended by the arc at the earth's center) connecting a DF site, any other point and the north pole. Such a triangle is sketched in Figure 3. The DF site is located at vertex B. Vertex C is at any other point. Vertex A is at the north pole. The arc c is the complement of the geocentric latitude of the DF site. Similarly, the arc b is the complement of the geocentric latitude of the other point. Angle A is equal to the difference in longitude between the DF site and the point. The dotted arc included on Figure 3 corresponds to a signal azimuth with angle indicated by θ .



Polar Triangle with Signal Azimuth

FIGURE 3

Angle A corresponds to the smaller of two possible polar triangles connecting any two points and the north pole. Visually, there is no problem in deciding which polar triangle to choose. However, when this triangle contains the prime meridian, the discontinuity at longitude $360 = \text{longitude } 0$ results in assigning the value of the vertex of the larger polar triangle when the longitude difference is taken. The mathematical model can be forced to correspond to the spatial relationships by adding the proviso that if the difference in degrees of longitude is greater than 180 degrees, then let A equal 360 degrees minus the difference. This restricts the size of angle A to one hemisphere. The absolute value of the difference can be taken in case the DF site is east of (greater longitude than) the point. These relationships are summarized below with subscript s designating the DF site and subscript t designating any test point.

$$c = 90 - \psi_s \quad (2-7)$$

$$b = 90 - \psi_t \quad (2-8)$$

$$A = \begin{cases} |\lambda_t - \lambda_s|, & \text{if } |\lambda_t - \lambda_s| \leq 180 \\ 360 - |\lambda_t - \lambda_s|, & \text{if } |\lambda_t - \lambda_s| > 180 \end{cases} \quad (2-9)$$

1. Point Azimuth Solution

The value of arc a connecting the DF site and test point is given by the law of cosines for sides,

$$\cos a = \cos b \cos c + \sin b \sin c \cos A \quad (2-10)$$

There is no quadrant ambiguity since the cosine takes negative values in the second quadrant and arc a is always less than or equal to 180 degrees because of the restriction on angle A to one hemisphere. Since the cosine of angle A is of interest in determining arc a , both the absolute value sign and the condition can be removed by applying a reduction formula. From equation (2-9),

$$\cos A = \begin{cases} \cos |\lambda_t - \lambda_s| \\ \cos (360 - |\lambda_t - \lambda_s|), \text{ if } |\lambda_t - \lambda_s| > 180 \end{cases}$$

But, through reduction,

$$\cos (360 - |\lambda_t - \lambda_s|) = \cos |\lambda_t - \lambda_s|$$

The absolute value sign can be dropped since $\cos (\lambda_t - \lambda_s) = \cos (\lambda_s - \lambda_t)$. From this result and equation (2-10)

$$\cos a = \cos b \cos c + \sin b \sin c \cos (\lambda_t - \lambda_s) \quad (2-11)$$

The vertex angle B at the DF site can be found by applying one of the laws of cosines for sides,

$$\cos B = \frac{\cos b - \cos c \cos a}{\sin c \sin a} \quad (2-12)$$

Equation (2-12) is valid for test points either east or west of the DF site at vertex B .

In order to make comparisons with a signal azimuth, angle B must be converted to an azimuth from true north, denoted β . Visually, the conversion is based on whether the test point is eastward or westward of the DF site with respect to the proper polar triangle. Mathematically, the

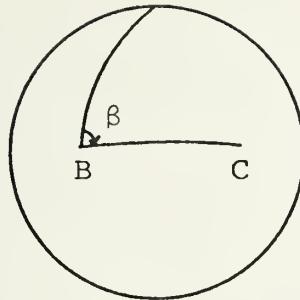
conversion is based on the magnitude of the longitudes of the DF site and the test point, but must also account for the discontinuity at the prime meridian. There are four general cases with two distinct outcomes as illustrated in Figure 4.

2. Point Azimuth and Signal Azimuth Difference

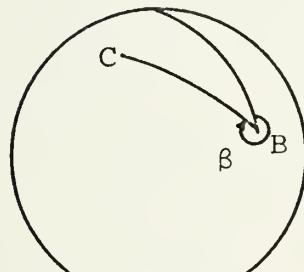
From a given DF site, the difference between the azimuth of any test point and a signal azimuth is the smaller of the two possible outcomes, depending upon whether the difference is measured clockwise or counter-clockwise. The azimuth difference is denoted ω , and the azimuth discontinuity at true north requires that the computation be conditional. With a clockwise rotation for the azimuths, the smaller angle of azimuth difference will be positive if β is leading and negative if β trails θ . The possible outcomes and their conditions are as follows.

- a. If $|\beta - \theta| < 180$, then $\omega = \beta - \theta$
 - b. If $|\beta - \theta| > 180$ and $\beta > \theta$, then $\omega = \beta - \theta - 360$
 - c. If $|\beta - \theta| > 180$ and $\beta < \theta$, then $\omega = \beta - \theta + 360$
- (2-14)

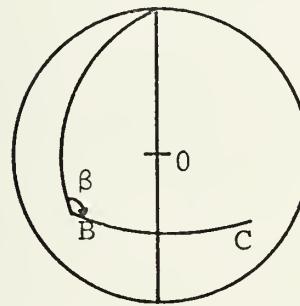
The sets of conditions on computing $\beta(B)$ in equation (2-13) and in computing $\omega(\beta, \theta)$ in equation (2-14) yield six possible outcomes for computing azimuth difference as a function of the vertex angle B and signal azimuth θ .



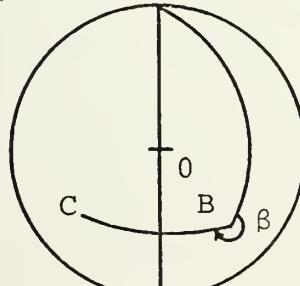
- a. $\lambda_s < \lambda_t$ and $\lambda_t - \lambda_s < 180$, then $\beta = B$



- b. $\lambda_s > \lambda_t$ and $\lambda_s - \lambda_t < 180$, then $\beta = 360 - B$



- c. $\lambda_s > \lambda_t$ and $\lambda_s - \lambda_t > 180$, then $\beta = B$



- d. $\lambda_s < \lambda_t$ and $\lambda_t - \lambda_s > 180$, then $\beta = 360 - B$

Azimuth and Vertex Angle

FIGURE 4

$$\omega = \begin{cases} B - \theta \\ B - 360 - \theta \\ B + 360 - \theta \\ -B + 360 - \theta \\ -B - \theta \\ -B + 720 - \theta \end{cases}$$

Two utility variables may be introduced in order to express azimuth difference in a constant form with values dependent upon the existing conditions.

$$\omega = pB + q - \theta \quad (2-15)$$

The pair of utility variables can take on any one of the following sets of values

$$(p,q) = \begin{cases} (1,0) \\ (1,-360) \\ (1,360) \\ (-1,360) \\ (-1,0) \\ (-1,720) \end{cases}$$

Concluding development, angle B can be expressed as a differentiable function of sines and cosines of known values of longitude and geocentric latitude. From equations (2-11) and (2-12) and the application of an angle-difference relation,

$$B = \cos^{-1} \left[\left(\cos b - \cos c [\cos b \cos c + \sin b \sin c (\cos \lambda_t \cos \lambda_s + \sin \lambda_t \sin \lambda_s)] \right) / \left[\sin c \sin \left(\cos^{-1} [\cos b \cos c + \sin b \sin c (\cos \lambda_t \cos \lambda_s + \sin \lambda_t \sin \lambda_s)] \right) \right] \right]$$

Substituting the values for arcs b and c from equations (2-7) and (2-8) and using the reduction formulas $\cos (90 - x) = \sin x$ and $\sin (90 - x) = \cos x$,

$$B = \cos^{-1} \left[\left(\sin \psi_t - \sin \psi_s [\sin \psi_t \sin \psi_s + \cos \psi_t \cos \psi_s (\cos \lambda_t \cos \lambda_s + \sin \lambda_t \sin \lambda_s)] \right) / \left[\cos \psi_s \sin \left(\cos^{-1} [\sin \psi_t \sin \psi_s + \cos \psi_t \cos \psi_s (\cos \lambda_t \cos \lambda_s + \sin \lambda_t \sin \lambda_s)] \right) \right] \right]$$

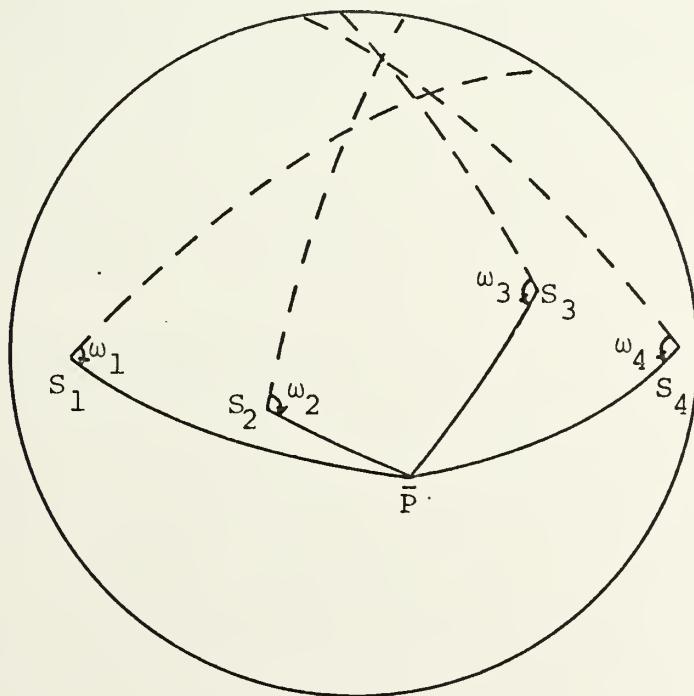
By multiplying the term $\sin \psi_s$ through the interior brackets of the numerator, writing the first two resulting terms as $\sin \psi_t (1 - \sin^2 \psi_s)$, which is equal to $\sin \psi_t \cos^2 \psi_s$, $\cos \psi_s$ can be cancelled from both numerator and denominator.

$$B = \cos^{-1} \left(\left[\sin \psi_t \cos \psi_s - \sin \psi_s \cos \psi_t (\cos \lambda_t \cos \lambda_s + \sin \lambda_t \sin \lambda_s) \right] / \left[\sin \left(\cos^{-1} [\sin \psi_t \sin \psi_s + \cos \psi_t \cos \psi_s (\cos \lambda_t \cos \lambda_s + \sin \lambda_t \sin \lambda_s)] \right) \right] \right) \quad (2-16)$$

III. THE FIX SOLUTION

In Figure 5, each of four DF sites have measured azimuths of a signal from the same emitter at an unknown location. The signal azimuths are plotted for illustration as dotted arcs. A test point, $\bar{P}(\lambda_t, \psi_t)$, is established, and azimuths from each DF site to \bar{P} are drawn in solid arcs. A better fit can be found by moving \bar{P} somewhere within the region bounded by the dotted arcs.

Given a number n of DF sites, each reporting the azimuth of a received signal with respect to its own location to a net control, the fix problem can be viewed as finding the



Azimuth Differences

FIGURE 5

point on the globe which minimizes the sum of some function of the azimuth differences ω_i , for $i = 1, 2, \dots, n$.

A. OBJECTIVE FUNCTION

Since the azimuth differences are signed, minimizing their sum may produce a solution that lies outside the region bounded by arcs corresponding to signal azimuths when large positive and large negative azimuth differences cancel each other out. Minimizing the sum of the absolute values of azimuth differences avoids this problem of positive and negative differences; however, since the partial derivatives of absolute value are not computationally tractable, solution techniques are limited. Minimizing the sum of the squared azimuth differences provides an objective function that is differentiable and reflects absolute magnitude. Therefore, the solution of the fix problem is taken to be that pair of coordinates (λ_t, ψ_t) which will

$$\text{Minimize } \sum_{i=1}^n \omega_i^2 (\lambda_t, \psi_t) \quad (3-1)$$

B. SECOND-ORDER GRADIENT SEARCH

Dropping the t subscript to simplify exposition, the objective function, equation (3-1), can be represented as $f(\lambda, \psi)$. Given an initial test point (λ, ψ) , a truncated Taylor's series for a function of two variables can be used to approximate the functional values of other test points $(\lambda + \Delta\lambda, \psi + \Delta\psi)$.

$$f(\lambda + \Delta\lambda, \psi + \Delta\psi) \doteq f(\lambda, \psi) + \Delta\psi \frac{\partial f(\lambda, \psi)}{\partial \psi} + \Delta\lambda \frac{\partial f(\lambda, \psi)}{\partial \lambda}$$

$$+ \frac{1}{2}(\Delta\lambda)^2 \frac{\partial^2 f(\lambda, \psi)}{\partial \lambda^2} + \Delta\lambda \Delta\psi \frac{\partial^2 f(\lambda, \psi)}{\partial \lambda \partial \psi} + \frac{1}{2}(\Delta\psi)^2 \frac{\partial^2 f(\lambda, \psi)}{\partial \psi^2}$$

Expressing the approximation in matrix form and letting superscript t indicate the transpose,

$$f \begin{pmatrix} \lambda + \Delta\lambda \\ \psi + \Delta\psi \end{pmatrix} \doteq f(\lambda, \psi)^t + (\Delta\lambda, \Delta\psi) \begin{pmatrix} \frac{\partial f(\lambda, \psi)}{\partial \lambda} \\ \frac{\partial f(\lambda, \psi)}{\partial \psi} \end{pmatrix}$$

$$+ \frac{1}{2}(\Delta\lambda, \Delta\psi) \begin{pmatrix} \frac{\partial^2 f(\lambda, \psi)}{\partial \lambda^2} & \frac{\partial^2 f(\lambda, \psi)}{\partial \lambda \partial \psi} \\ \frac{\partial^2 f(\lambda, \psi)}{\partial \lambda \partial \psi} & \frac{\partial^2 f(\lambda, \psi)}{\partial \psi^2} \end{pmatrix} \begin{pmatrix} \Delta\lambda \\ \Delta\psi \end{pmatrix}$$

Letting \bar{P}_j denote an initial test point $(\lambda, \psi)^t$, letting \bar{R} denote the vector $(\Delta\lambda, \Delta\psi)^t$, letting ∇ denote the gradient of $f(\bar{P}_j)$ and letting ∇^2 denote the Hessian matrix of f at \bar{P}_j , the approximation can be expressed as

$$f(\bar{P}_j + \bar{R}) \doteq f(\bar{P}_j) + \bar{R}^t \nabla f(\bar{P}_j) + \frac{1}{2} \bar{R}^t \nabla^2 f(\bar{P}_j) \bar{R}$$

Then,

$$f(\bar{P}_j + \bar{R}) - f(\bar{P}_j) \doteq \bar{R}^t \nabla f(\bar{P}_j) + \frac{1}{2} \bar{R}^t \nabla^2 f(\bar{P}_j) \bar{R} \quad (3-2)$$

Starting from an initial test point at \bar{P}_j , the search procedure can be understood as a technique to find a vector \bar{R} such that $f(\bar{P}_j + \bar{R})$ is the minimum of the objective function. Such a vector \bar{R} also minimizes the left hand side of equation (3-2). Since the quadratic form on the right hand side is an

approximation of the value of the left hand side, it must be near a minimum value when the left hand side is minimized. The approximation works both ways, so that a vector \bar{R} which minimizes the right hand side must approximately minimize the left hand side. The right hand side is in a form that yields to analysis of its extreme values with respect to \bar{R} at a fixed \bar{P}_j . One of the conditions necessary for the quadratic form to be at a minimum is that its gradient be equal to zero. Therefore, it is necessary to find a vector \bar{R} which fulfills this condition. Taking the gradient of equation (3-2) with respect to \bar{R} ,

$$\begin{aligned} \nabla_{\bar{R}} & \left[\bar{R}^T \nabla f(\bar{P}_j) + \frac{1}{2} \bar{R}^T \nabla^2 f(\bar{P}_j) \bar{R} \right] \\ &= \nabla_{\bar{R}} \left[\bar{R}^T \nabla f(\bar{P}_j) \right] + \nabla_{\bar{R}} \left[\frac{1}{2} \bar{R}^T \nabla^2 f(\bar{P}_j) \bar{R} \right] \\ &= \nabla f(\bar{P}_j) + \frac{1}{2} \nabla^2 f(\bar{P}_j) \bar{R} + \frac{1}{2} \bar{R}^T \nabla^2 f(\bar{P}_j) \\ &= \nabla f(\bar{P}_j) + \nabla^2 f(\bar{P}_j) \bar{R} \end{aligned}$$

Setting the gradient of the quadratic form at \bar{R} equal to zero,

$$\nabla f(\bar{P}_j) + \nabla^2 f(\bar{P}_j) \bar{R} = 0$$

Then,

$$\nabla^2 f(\bar{P}_j) \bar{R} = -\nabla f(\bar{P}_j)$$

And, providing that $\nabla^2 f(\bar{P}_j)$ is invertible,

$$\bar{R} = -\left[\nabla^2 f(\bar{P}_j) \right]^{-1} \nabla f(\bar{P}_j) \quad (3-3)$$

\bar{R} specifies both a direction and magnitude in which to move toward an extreme value of the objective function.

This consequence of the Taylor approximation can be improved by applying a scalar coefficient, denoted δ , to \bar{R} in order to adjust the magnitude and, if \bar{R} points to a maximum value, the direction. The next test point is then given by

$$\bar{P}_{j+1} = \bar{P}_j + \delta \bar{R} \quad (3-4)$$

Combining equations (3-3) and (3-4),

$$\bar{P}_{j+1} = \bar{P}_j - \delta [\nabla^2 f(\bar{P}_j)]^{-1} \nabla f(\bar{P}_j) \quad (3-5)$$

The search procedure can be terminated when further improvements are of little consequence, most readily determined by comparing the distance between locations on successive iterations.

In order to solve equation (3-5), it is necessary to compute the inverse Hessian of the objective function.

$$[\nabla^2 f(\bar{P})]^{-1} = \frac{1}{\det[\nabla^2 f(\bar{P})]} \begin{pmatrix} \frac{\partial^2 f(\bar{P})}{\partial \psi^2} & -\frac{\partial^2 f(\bar{P})}{\partial \lambda \partial \psi} \\ -\frac{\partial^2 f(\bar{P})}{\partial \lambda \partial \psi} & \frac{\partial^2 f(\bar{P})}{\partial \lambda^2} \end{pmatrix}$$

where the determinant of the Hessian

$$\det[\nabla^2 f(\bar{P})] = \frac{\partial^2 f(\bar{P})}{\partial \lambda^2} \frac{\partial^2 f(\bar{P})}{\partial \psi^2} - \left[\frac{\partial^2 f(\bar{P})}{\partial \lambda \partial \psi} \right]^2$$

Substituting the inverse Hessian in equation (3-5) and expanding the equation by matrix multiplication of the inverse Hessian times the gradient yields a new test point with δ the only unknown.

$$\begin{aligned}
 & \left(\frac{\frac{\partial^2 f(\bar{P}_j)}{\partial \psi^2} \frac{\partial f(\bar{P}_j)}{\partial \lambda} - \frac{\partial^2 f(\bar{P}_j)}{\partial \lambda \partial \psi} \frac{\partial f(\bar{P}_j)}{\partial \psi}}{\frac{\partial^2 f(\bar{P}_j)}{\partial \lambda^2} \frac{\partial^2 f(\bar{P}_j)}{\partial \psi^2} - \left(\frac{\partial^2 f(\bar{P}_j)}{\partial \lambda \partial \psi} \right)^2}, \right. \\
 & \quad \left. \frac{\frac{\partial^2 f(\bar{P}_j)}{\partial \lambda^2} \frac{\partial f(\bar{P}_j)}{\partial \psi} - \frac{\partial^2 f(\bar{P}_j)}{\partial \lambda \partial \psi} \frac{\partial f(\bar{P}_j)}{\partial \lambda}}{\frac{\partial^2 f(\bar{P}_j)}{\partial \lambda^2} \frac{\partial^2 f(\bar{P}_j)}{\partial \psi^2} - \left(\frac{\partial^2 f(\bar{P}_j)}{\partial \lambda \partial \psi} \right)^2} \right) \\
 \bar{P}_{j+1} &= \bar{P}_j - \delta
 \end{aligned} \tag{3-6}$$

The solution of equation (3-6) in turn requires the calculation of all first and second partial derivatives of the objective function.

$$\begin{aligned}
 \frac{\partial f(\bar{P})}{\partial \lambda} &= \frac{\partial \left(\sum_{i=1}^n \omega_i^2 \right)}{\partial \lambda} \\
 &= \sum_{i=1}^n 2p_i(p_i B_i + q_i - \theta_i) \frac{\partial B_i}{\partial \lambda}
 \end{aligned} \tag{3-7}$$

Similarly,

$$\frac{\partial f(\bar{P})}{\partial \psi} = \sum_{i=1}^n 2p_i(p_i B_i + q_i - \theta_i) \frac{\partial B_i}{\partial \psi} \tag{3-8}$$

$$\frac{\partial^2 f(\bar{P})}{\partial \lambda^2} = \sum_{i=1}^n \left[2p_i^2 \left(\frac{\partial B_i}{\partial \lambda} \right)^2 + 2p_i (p_i B_i + q_i - \theta_i) \frac{\partial^2 B_i}{\partial \lambda^2} \right] \tag{3-9}$$

$$\frac{\partial^2 f(\bar{P})}{\partial \psi^2} = \sum_{i=1}^n \left[2p_i^2 \left(\frac{\partial B_i}{\partial \psi} \right)^2 + 2p_i (p_i B_i + q_i - \theta_i) \frac{\partial^2 B_i}{\partial \psi^2} \right] \tag{3-10}$$

$$\frac{\partial^2 f(\bar{P})}{\partial \lambda \partial \psi} = \sum_{i=1}^n \left[2p_i^2 \frac{\partial B_i}{\partial \lambda} \frac{\partial B_i}{\partial \psi} + 2p_i (p_i B_i + q_i - \theta_i) \frac{\partial^2 B_i}{\partial \lambda \partial \psi} \right] \tag{3-11}$$

Solving equations (3-7) through (3-11) requires the lengthy procedure of taking all first and second partial derivatives of equation (2-16), the expression for angle B. Solutions to these equations are contained in the FORTRAN Fix Program included in this paper.

C. GOLDEN SECTION SEARCH

From equations (2-15) and (3-1), the objective function can be written,

$$\text{Minimize } \sum_{i=1}^n (p_i B_i + q_i - \theta_i)^2, \quad (3-12)$$

the sum of the squared azimuth differences.

Letting x denote the indicated move in longitude and y denote the indicated move in latitude given by equation (3-6),

$$\begin{aligned} \bar{R} &= (\Delta\lambda, \Delta\psi)^T \\ &= (x, y)^T \end{aligned}$$

Then angle B_i , given in equation (2-16), becomes at \bar{P}_{j+1}

$$B_i = \cos^{-1} \left[\left(\begin{array}{c} \sin(\psi_j - \delta y) \cos\psi_i - \sin\psi_i \cos(\psi_j - \delta y) \\ \cos(\lambda_j - \delta x) \cos\lambda_i + \sin(\lambda_j - \delta x) \sin\lambda_i \end{array} \right) \middle/ \left(\begin{array}{c} \sin \left[\cos^{-1} \right. \\ \left. \begin{array}{c} \sin(\psi_j - \delta y) \sin\psi_i + \cos(\psi_j - \delta y) \cos\psi_i [\cos(\lambda_j - \delta x) \\ \cos\lambda_i + \sin(\lambda_j - \delta x) \sin\lambda_i] \end{array} \right] \end{array} \right) \right] \quad (3-13)$$

where subscript j denotes test point coordinates and subscript i denotes DF site coordinates. Since test point

coordinates are arbitrarily known and DF site coordinates are constant for the problem, the substitution of equation (3-13) into equation (3-12), the objective function, results in the problem of minimizing a function with a single unknown, δ , the scalar coefficient.

$$\text{Minimize } f(\delta) = \sum_{i=1}^n \omega_i^2(\delta) \quad (3-14)$$

The objective function could be evaluated at incremental values of δ , say over the interval (a, b) . Then that δ which yields the smallest functional value would be chosen as the scalar coefficient to adjust the move indicated by the second-order gradient search. The Golden Section Search (reference 1) provides a fast and methodical approach to find this minimizing δ . Figure 6 illustrates the problem; however, the actual curve representing $f(\delta)$ is unknown at the outset of the problem.

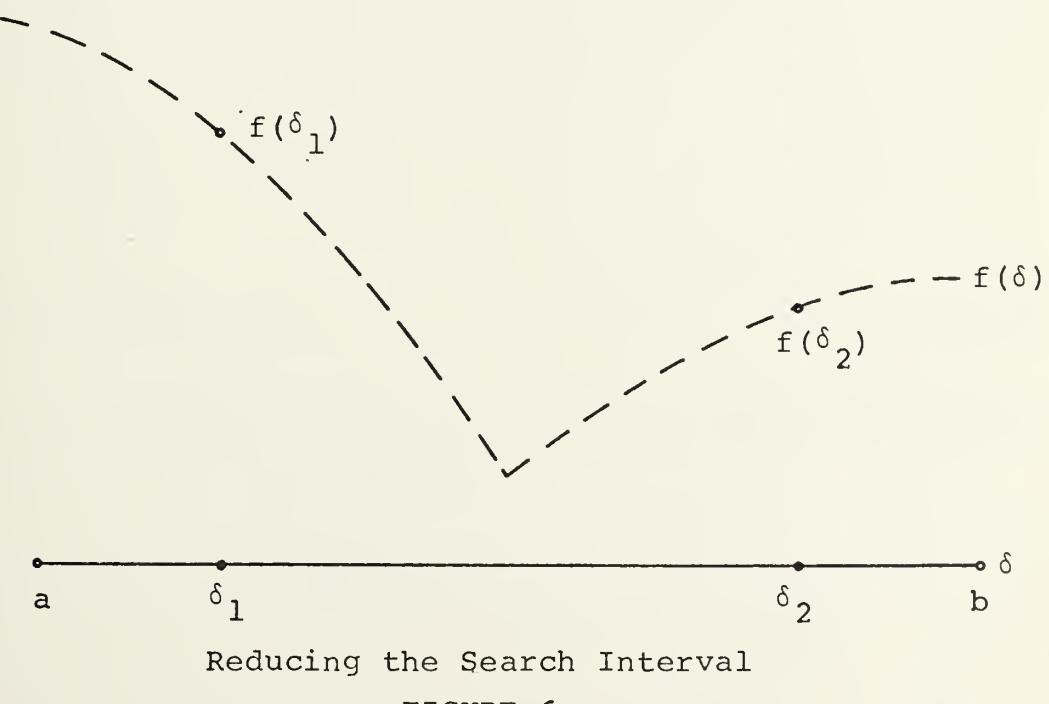


FIGURE 6

Based on the assumption that $f(\delta)$ has a local minimum on (a,b) , the objective is to find that minimum without evaluating the function at too many points in (a,b) . The Golden Section Search systematically eliminates portions of the interval by comparing functional values of pairs of points in (a,b) . From Figure 6, $f(\delta_2)$ is less than $f(\delta_1)$; therefore, the portion (a,δ_1) can be eliminated in searching for the minimum, reducing the search interval to (δ_1,b) . A new point then is selected for comparison with $f(\delta_2)$, resulting in a second elimination of whichever end interval corresponds to the higher functional value.

Letting I_n represent the length of the n^{th} search interval,

$$I_1 = b - a$$

Two different values of δ are tested at each trial. By letting the two values be symmetric on I_n , the length of I_{n+1} is the same regardless of which end interval is dropped. Letting σ denote the relative position of δ_1 in I_1 ,

$$\sigma = (\delta_1 - a) / (b - a)$$

Then

$$\delta_1 = \sigma(b - a) + a \quad (3-15)$$

Since δ_2 is symmetric with δ_1 , the relative position of δ_2 in I_1 is given by $(1 - \sigma)$, so that

$$1 - \sigma = (\delta_2 - a) / (b - a),$$

and

$$\delta_2 = (1 - \sigma)(b - a) + a \quad (3-16)$$

Provided that δ_2 yielded the minimum functional value, as in Figure 6, then δ_2 becomes the first element of the symmetric pair in I_2 and its relative position in I_2 is given by

$$\begin{aligned} \sigma &= (\delta_2 - \delta_1)/(1 - \sigma)(b - a) \\ &= [(1 - \sigma)(b - a) + a - \delta_1]/(1 - \sigma)(b - a) \end{aligned}$$

from equation (3-16), so that

$$\delta_1 = (1 - \sigma)^2(b - a) + a \quad (3-17)$$

From equations (3-15) and (3-17), it follows that

$$\begin{aligned} \sigma(b - a) + a &= (1 - \sigma)^2(b - a) + a \\ \sigma &= (1 - \sigma)^2 \end{aligned} \quad (3-18)$$

If, on the other hand, δ_1 yields the minimum functional value, then δ_1 becomes the second element of the symmetric test pair in I_2 , and its relative position in I_2 is given by

$$1 - \sigma = (\delta_1 - a)/(1 - \sigma)(b - a)$$

so that

$$\delta_1 = (1 - \sigma)^2(b - a) + a,$$

as in equation (3-17), and leading to the same result as in equation (3-18).

From equation (3-18)

$$\sigma^2 - 3\sigma + 1 = 0$$

which can be solved by the quadratic formula, so that

$$\sigma = \left[3 \pm (9 - 4)^{\frac{1}{2}} \right] / 2$$

$$= 0.382$$

Substituting this result in equations (3-15) and (3-16),

$$\delta_1 = 0.382 (b - a) + a \quad (3-19)$$

$$\delta_2 = 0.618 (b - a) + a \quad (3-20)$$

Generalizing, let b_n and a_n represent the endpoints of the n^{th} search interval. Then

$$\delta_1 = 0.382 (b_n - a_n) + a_n \quad (3-21)$$

$$\delta_2 = 0.618 (b_n - a_n) + a_n \quad (3-22)$$

After the first comparison on I_1 , either δ_1 or δ_2 is the survivor from the preceding iteration, so that a comparison can be made after evaluating only the new test point symmetric with the survivor. The Golden Section Search terminates when $b_n - a_n < \epsilon$, where ϵ may be tailored to the magnitude of the indicated move.

The value of δ which minimizes $f(\delta)$ over (a_1, b_1) is entered in equation (3-6), resulting in a new test point

$$\bar{P}_{j+1} = (\lambda_j, \psi_j)^t + \delta(x, y)^t \quad (3-23)$$

$$= \bar{P}_j + \delta \bar{R}$$

D. FORTRAN FIX PROGRAM

The equations presented in this paper were coded in FORTRAN. The resulting program has been successfully

executed on an IBM 360/67 computer system in the W. R. Church Computer Center at the Naval Postgraduate School.

1. Program Description

The program contains an identification number, longitude, latitude and name for each DF site in an imaginary network. Inputs to the program are DF site identification number and signal azimuth. After DF site names and coordinates are correlated with input site identification numbers, necessary trigonometric functions of site coordinates are computed and stored in an array. Searching for the minimum sum of azimuth differences begins from (0,0) with a Golden Section Search on longitude using (360,0) as the indicated move. The result is a minimizing longitude on the equator from which a Golden Section Search is conducted on geocentric latitude with indicated move (0,180). From this minimizing geocentric latitude, another Golden Section Search is conducted on longitude with (180,0) as the indicated move. Such alternating searches on longitude and geocentric latitude continue until diminishing returns are indicated by a new position being less than 500 nautical miles from the previous minimizing position. The initial interval for the Golden Section Search is set at (-1,1), the endpoints for equations (3-19) and (3-20). Subsequent endpoints are modified by the search results, producing values for equations (3-21) and (3-22).

The program then executes second-order gradient searches refined by Golden Section Searches on indicated

moves until the most recent move is less than some input termination value, say one-tenth of a nautical mile. Indicated moves, given by equation (3-6), are unbounded and might occasionally go off the coordinate system of Figure 1. Scaling any such wild moves to the coordinate system by modulo arithmetic destroys both the direction and magnitude of the vector. The indicated moves are kept in bounds by conversion to a unit vector in radians until such moves are of magnitude less than 30 nautical miles, such a small step indicating that the solution is at hand. By converting the indicated move to a unit vector prior to refinement by a scalar coefficient from the Golden Section Search, moves from the second-order gradient search are limited to about 3440 nautical miles per iteration.

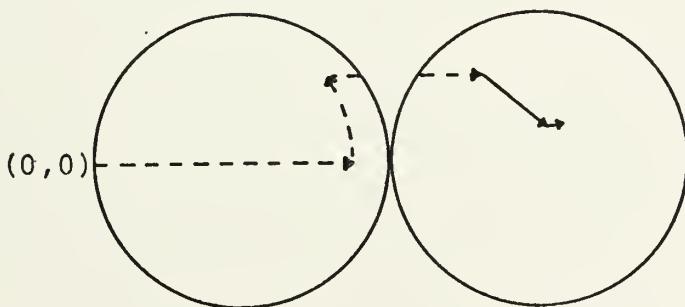
Search output on successive iterations is optional, depending on an operator entry. Unless search progress is requested, searching terminates after thirty iterations even if the latest step is greater than the input termination value. After every tenth second-order gradient search, if such a high number should be required, the search effort is jolted by conducting alternate Golden Section Searches on longitude with indicated move (90,0) and latitude with indicated move (0,90). Such Golden Section Search interruptions only yield a new test point when they reduce the current value of the objective function.

Prior to solution output, geocentric latitude is converted to geographical latitude using equation (2-6), and

the solution is given in degrees and minutes of longitude, east or west, and latitude, north or south.

2. Results

The program has worked successfully in over ten thousand runs against simulated targets spread randomly over the globe with various networks of DF sites and various numbers of sites intercepting a given target. In actual practice, prior information about a target's area of operation might produce a better starting point than $(0,0)$; however, the time required to type such starting coordinates on a keyboard for input would exceed the total time required for the solution starting from $(0,0)$, which averages less than a second and includes some diagnostic output useful in analyzing results. Figure 7 illustrates a representative set of search iterations with Golden Section Searches plotted as dashed lines and second-order gradient searches solid.



Searching

FIGURE 7

IV. TARGET LOCATION CONFIDENCE REGION

That location yielding the minimum sum of squared azimuth differences, or least squares solution, would coincide with the location of the signal emitter if all measured signal azimuths with respect to the DF sites were free from error. Since then all great circle arcs corresponding to signal azimuths would intersect at a point, the minimum value of the objective function would be equal to zero, as is always the case when only two signal azimuths are available to compute a fix. Because the sites in a DF network do not produce errorless measurements, solutions rarely coincide with the emitter location, and there is the question of how close the emitter might be to the least squares solution.

A. SIGNAL AZIMUTH ERROR DISTRIBUTIONS

Supposing that signal azimuth measurements made at individual DF sites are not biased in any of the possible directions, then it follows that signed errors have a mean value of zero. If such measurements were found to be biased in certain directions during routine calibration trials, then the amount of bias could be applied as a correction to subsequent measurements, producing an estimated mean value of zero. Lacking any reason to assume a different distribution and following the lead given in the literature of direction finding, the error distribution can be taken to

be normally distributed with mean zero and variance σ_i^2 , assumed known and independent from site to site.

The error distribution can be made to be symmetric around zero by following the sign conversion that clockwise errors are negative and counter-clockwise errors are positive in sign.

B. THE OBJECTIVE FUNCTION VALUE AS A CHI-SQUARE STATISTIC

From standard textbooks on probability theory (for example, reference 2), it can be verified that if a random variable is normally distributed with mean zero and variance σ^2 , then the square of that random variable is governed by the Gamma distribution with parameters $(1/2, 1/2\sigma^2)$. Furthermore, the sum of n independent Gamma distributed random variables divided by the variance of the underlying normal distributions is a Chi-Square statistic with n degrees of freedom. Hence, where ω denotes the azimuth difference,

$$\omega_i \sim N(0, \sigma_i^2) \quad (4-1)$$

$$\omega_i^2 \sim \Gamma\left(\frac{1}{2}, \frac{1}{2\sigma_i^2}\right) \quad (4-2)$$

$$\sum_{i=1}^n \frac{1}{\sigma_i^2} \omega_i^2 \sim \chi^2_{(n)} \quad (4-3)$$

Since the normal distribution here referred to applies to the azimuth differences between the measured signal azimuth and the azimuth of the unknown emitter location, the ω_i refer to actual and unknown errors rather than calculated differences between solution azimuths and signal azimuths.

Equation (4-3) therefore specifies the distribution of values of the objective function (equation (3-12)) at unknown emitter locations under the assumption that azimuth errors are normally distributed with known variance.

C. CONFIDENCE LEVELS AND REGION BOUNDARIES

Since the sum of squared azimuth errors divided by variances is a Chi-Square statistic under the previous assumptions, there is a specific probability, denoted α , that this sum will be less than or equal to any given value. Equivalently, for a fixed probability α , or confidence level, there corresponds an upper bound, denoted $C_{(n,\alpha)}$, which depends upon the degrees of freedom as well as the confidence level.

$$P \left(\sum_{i=1}^n \frac{1}{\sigma_i^2} \omega_i^2 \leq C_{(n,\alpha)} \right) = \alpha \quad (4-4)$$

The choice of confidence level, α , is arbitrary. The corresponding value for $C_{(n,\alpha)}$ can be found in Chi-Square tables.

For a given fix problem, the value of the sum of least squares, denoted f^* , is the minimum possible value of the Chi-Square statistic for the observed signal azimuths. Furthermore, by evaluating the objective function at successive points outward in any direction from the least squares solution, it can be seen that the farther away the point evaluated up to the antipode, the higher the objective function value. The location corresponding to f^* can

therefore be regarded as a point in a region which contains all the possible emitter locations for which the value of the objective function is less than or equal to an upper bound $C_{(n,\alpha)}$, given the set of observed signal azimuths. Such a region is bounded by all locations for which the objective function has a value equal to $C_{(n,\alpha)}$, and it can be stated that, for the observed signal azimuths, there is a probability α that the bounded region will contain the unknown emitter location.

Daniels and Vajda (1951) challenge the utility of such an approach on the grounds that at given confidence levels the corresponding region may not exist with positive probability if n is greater than two. Their observation is a consequence of the fact that except for n equal to two all of the great circle arcs corresponding to signal azimuths seldom intersect at a single point thereby yielding f^* equal to zero.

Prior to the availability of the high-speed digital computer, the practical problems of producing such a region were certainly more striking than theoretical objections. Even so, the Daniels/Vajda criticism was more than casual, which may indicate a requirement for additional arguments to justify the use of such a confidence region.

Suppose that f^* is equal to k for some least squares solution, where

$$P \left(\sum_{i=1}^n \frac{1}{\sigma_i^2} \omega_i^2 \leq k \right) = p, \quad (4-5)$$

and the probability p may be determined by interpolating with the value k in a table of Chi-Squared values. Daniels and Vajda are concerned that if the confidence level α had been initially specified at some value less than p , then there exists no region of locations having objective function values less than or equal to k , k being the minimum possible value on the globe for the given signal azimuth observations. There are at least two responses to such an occurrence and the accompanying empty region objection: (1) Increase the confidence level whenever the region does not exist at the prior confidence level so that $C_{(n,\alpha)}$ is greater than k , and (2) Routinely specify a confidence level of sufficient magnitude to make the occurrence of an empty region highly improbable.

Either of the above alternatives insures the production of a non-empty confidence region. There remains the question of what is represented by a possible contention inferable from equation (4-5), namely that positive probability p is attached to a single point, the least squares solution. Any such statement would be puzzling when approached with the viewpoint that specific random variables on a continuum have zero probability and that probability statements can only be made about boundaries for continuous random variables such as location coordinates.

An answer may be attempted by first pointing out that azimuth errors are unknown and independent random variables, but they are attached to actual and unknown emitter azimuths

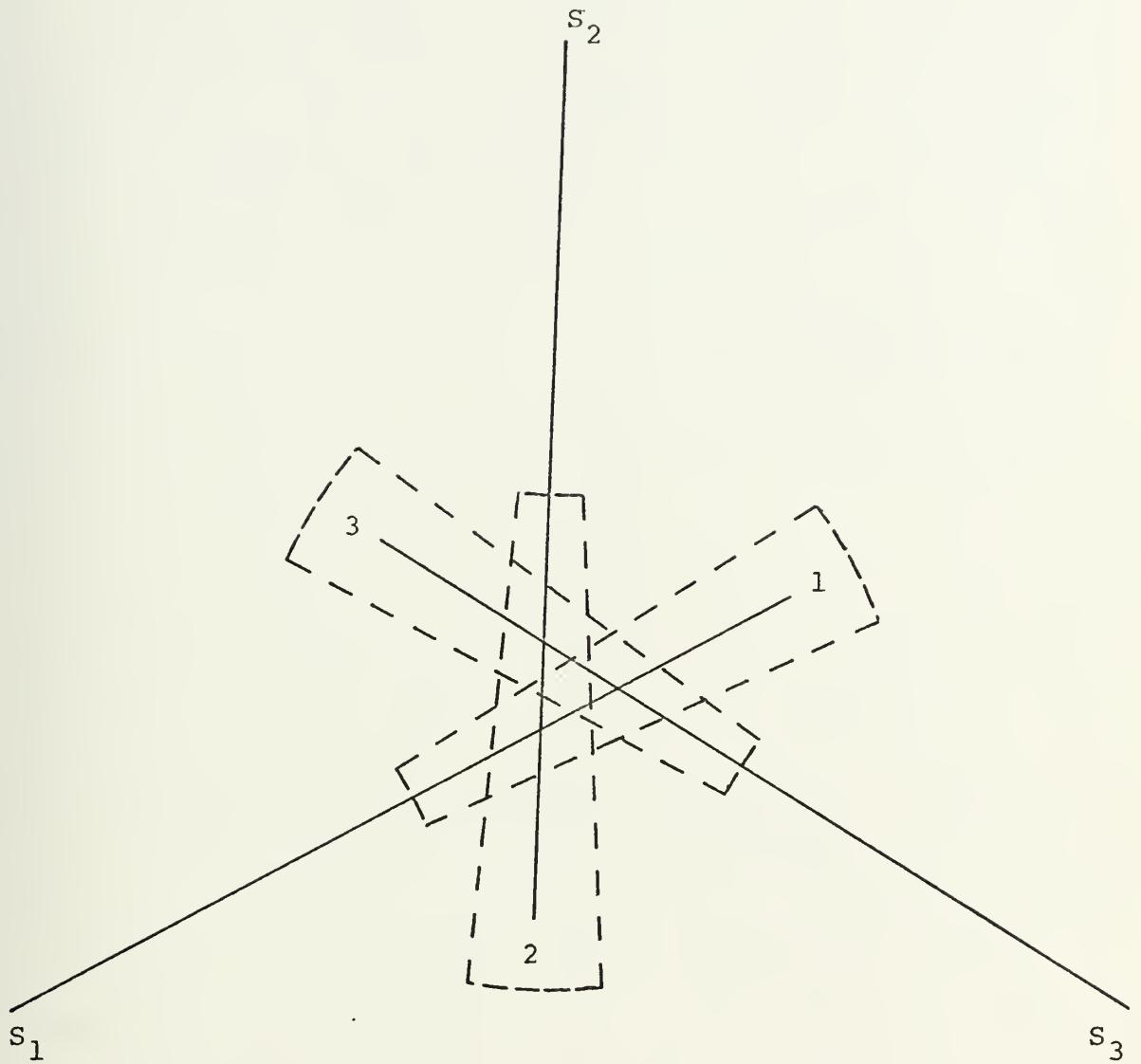
with respect to the set of intercepting DF sites. Observed signal azimuths have two components, the first of which, emitter azimuth, is dependent upon emitter location. The second component, azimuth error, is independent from site to site and is the theoretical basis for the Chi-Square contour confidence region. Prior to the production of a particular confidence region, the normality assumption permits the specification of upper bounds for the Chi-Square statistic corresponding to various levels of confidence. During production of the particular confidence region, the requirement that a single location must satisfy the observations places a lower bound on the Chi-Square statistic.

Each emitter azimuth can depart from the observed signal azimuth, but there are conceptual bands (based on the error distribution assumptions) bounding such departures for any given probability. When more than two such bands are overlapped in the solution region, then a proportion of the possible combinations of such departures are ruled out by the constraint that azimuth errors are attached to emitter azimuths. Whenever f^* is greater than zero, then the probability, denoted p^* , associated with f^* as an upper boundary from equation (4-5) represents the proportion of impossible sums of squares out of those permitted by the density functions governing azimuth errors.

A thought experiment may clarify these remarks. Suppose it were possible to specify all combinations of azimuth errors permitted by their respective density functions prior

to finding the least squares solution. There exists a location satisfying one of the postulated combinations of azimuth errors for a specific set of signal azimuths only if application of each of the postulated azimuth errors in the combination to its corresponding signal azimuth yields a corrected set of emitter azimuths whose corresponding great circle arcs intersect at a point. Few such postulated combinations in the infinite set can be satisfied since the point of intersection for any two corrected emitter azimuths uniquely determines the required azimuth corrections for the rest of the observations. Furthermore, there cannot exist a satisfying location for any postulated combination of azimuth errors having a sum of squares less than f^* since no such corrections are of sufficient magnitude to result in a common intersection.

These suggestions can be illustrated by an extreme and improbable example of emptiness of the confidence region for three signal azimuth great circle arcs forming a cocked hat. Suppose that the relationships between the DF sites and the emitter are as given in Figure 8, and suppose that the errors associated with the three DF sites are all governed by the standard normal distribution (variance equal to one degree squared). Suppose also that each of the three signal azimuths were in error by minus two degrees (where the probability that azimuth error is less than or equal to minus two degrees is equal to .0227) in order to form one of the two largest possible cocked hats for $|\omega_i|$ equal to two degrees. Then the least squares solution will be found



Bounded Signal Azimuths

FIGURE 8

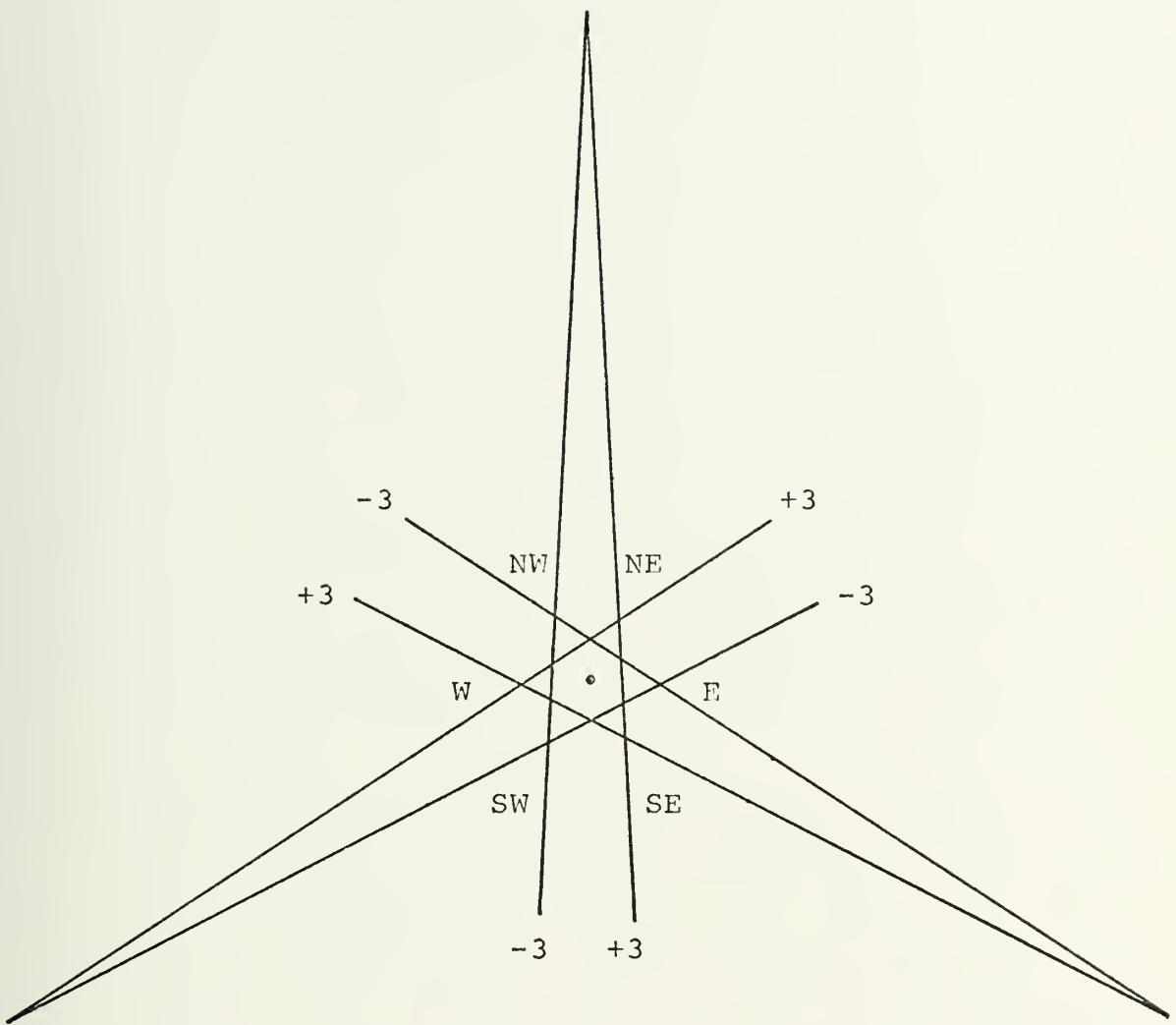
near the center of the cocked hat, and f^* is about equal to twelve degrees squared. For three degrees of freedom, there was a .992 probability that the Chi-Square statistic would be less than or equal to twelve degrees squared, indicating an empty confidence region at the .992 level (or any lower levels). Now the probability that $|\omega_i|$ is greater than four degrees is equal to .000, suggesting that a boundary of four degrees on either side of each signal azimuth great circle arc should contain all possible emitter azimuths from which each signal azimuth represents a departure. The situation is illustrated in Figure 8.

With each emitter azimuth free to be anywhere within its dashed band by the normality assumption, but dependent upon the actual emitter location and therefore constrained by the other dashed bands, the shaded intersection of the three bands must (for any non-negligible probability) contain the emitter. Moreover, the Chi-Square contour methodology would produce a circular confidence region within the shaded triangle for any confidence level greater than .992. The center of the shaded region corresponds to the least squares solution. Suppose a "center" to be the least squares solution united with all points forming a neighborhood within radius ϵ of the center. Then this "center" had a probability of $.992 + f(\epsilon)$ of containing the emitter location. Thus there was a probability of $.008 - f(\epsilon)$ of finding the emitter within a region bounded by a "centerless" circle inscribed in the shaded triangle. There is zero probability that the

emitter lies outside such an inscribed circle, or confidence region at the 100 percent level. The description of the confidence region as circular applies to Figure 8 only because the DF sites are equidistant and symmetric around the emitter, and because azimuth errors are minus two degrees.

The implication of the foregoing is that if errors are normally distributed with known variance, then abnormally large cocked hats would be indicative of excellent solutions via the least squares approach. The point would be difficult for operators to accept since high confidence is normally associated with a set of great circle arcs corresponding to signal azimuths that nearly intersect at a point, and large departures are suspect. The discrepancy between the above conclusion and experience may be attributed to the fact that abnormally large cocked hats only occur when outliers (signal azimuth errors that do not obey the empirically estimated distribution) enter the data base.

Sometimes strange relationships between the sizes of cocked hats and accuracy of locations (least squares solutions) can also be demonstrated with a network of three DF sites with errors governed by a Bernoulli distribution, having equal probability that azimuth error is plus or minus three degrees. Figure 9 illustrates the situation. Given an emitter at the center of the star, observed signal azimuths are in error by plus or minus three degrees, each with probability 1/2. All possible configurations of cocked hats are therefore represented on the figure for the



Bernoulli Random Error

FIGURE 9

sketched relationship between DF sites and emitter location. Six of the possible cocked hats are the small triangles at the tips of the star. The other two possibilities are the union of the body of the star with the small triangles at points NE, SE and W, and the union of the body of the star with the small triangles at points E, SW and NW. The latter two cocked hats are abnormally large and have f^* equal to 27 degrees squared (which is over twice the value of a Chi-Square statistic at the .995 level for a normal error distribution). Yet, the center of these largest cocked hats coincides approximately with the emitter location, indicating excellent least squares solutions, and the centers of the triangles at the points of the star, the smaller cocked hats, are far from the emitter location, even though their f^* value is relatively small.

The probability of occurrence of such large cocked hats when error is governed by a normal rather than Bernoulli distribution is negligible, and they do not appear when data is simulated with a normal error distribution having reasonable variance. However, in about 400 simulations during development of the Fix Program they were forced to appear by governing error with such a Bernoulli distribution. As would be expected from Figure 9, a plot of location error versus f^* verified that location error decreases as f^* approaches its maximum of 27 degrees squared for this unusual error distribution. Such a simulation naturally yields about 25 percent of the abnormally large cocked hats and 75 per-

cent triangles at the points of the star. With a random distribution of DF sites and signal emitters over the globe, the stars are not so well-shaped as that in Figure 9.

If one regards the fix problem with least squares solution and confidence region construction at confidence level α as a Bernoulli trial with success if the confidence region contains the emitter location and failure if the emitter is outside the confidence region, then the expected number of successes for m independent trials is αm . That the Chi-Square contour confidence regions have normalized frequency of containment about equal to the probability of containment is therefore expected to be true and was demonstrated to be true by simulation. At confidence level α , on about α percent of the trials non-empty confidence regions are produced which contain the actual emitter. On the remaining $(1 - \alpha)$ percent of the trials, outcomes are distributed among empty regions and non-empty regions with the actual emitter outside the confidence region. The lower the confidence level the higher the proportion of empty regions among the failures.

D. FORTRAN CONFIDENCE REGION PROGRAM

A plot subroutine is included in the Fix Program of this paper. Sample plots are included in the Computer Output section of the paper. The plot is a square grid of constant dimension, scaled in nautical miles to contain the confidence region. An asterisk on the grid indicates that the corresponding point has an objective function value greater than the Chi-Square boundary value for the specified confidence

level. The least squares solution is at the center of the grid and is indicated by the letter S. During program development, it was found that a 90 percent confidence level is sufficient to preclude the occurrence of empty confidence regions, and the sample regions are of that size.

1. Program Description

Given in a data statement of the subroutine are Chi-Square values corresponding to the numbers of possible degrees of freedom, up to n for a network of n DF sites, and the selected confidence level. The confidence region boundary value is selected according to the number of signal azimuths available to find a solution.

The dimension of the grid is 41 rows and 51 columns, tailored to the line printer distances between rows and columns, so that the ratio of the distances between rows and columns is 4:5. Depending upon the scale of the grid, coordinates on the globe are calculated to match each of the 2091 points on the grid. Curvature of the earth calculations are used to determine distances between points on the grid, thereby avoiding distortion inherent in planar assumptions.

At each of the points on a row, the sum of squared azimuth differences is computed for comparison with $C_{(n,\alpha)}$, an asterisk being assigned for grid entry if $C_{(n,\alpha)}$ is exceeded. After the confidence region is exited on a given row, remaining points on the row are assigned asterisks without computation. Similarly, after the confidence region

is completed, the rest of the grid is filled with asterisks without computation.

The area of the confidence region is estimated by counting the blanks and multiplying their sum by the area represented by each cell, 80 square miles in the case of a 400 by 400 nautical mile grid. If computations on the first row yield any points with objective function value less than or equal to $C_{(n,\alpha)}$, then the grid is rescaled to cover a greater area, and the computation starts over. If the confidence region leaks out on either of the sides of the grid during computation, then again the grid is rescaled, and computation starts over. In practice, the initial area covered by the grid can be tailored to the network and expected size of confidence regions.

2. Results

If the solution search in the Fix Program should terminate in the region of the least squares solution before the least squares point is found, then the confidence region is noticeably asymmetric about the center at S. The lack of symmetry indicates that the fix solution should be recomputed with a smaller termination step. The center of the confidence region, or least squares location, could also be estimated from the grid itself, but only to an accuracy limited by the dimensions of a cell.

The confidence region subroutine was used successfully in over a thousand runs with simulated data. Its chief deficiency is the computer time required to produce the grid

entries, averaging slightly over ten seconds for numbers of signal azimuths per target varying uniformly from three to nine. On a slower machine than the IBM 360/67, the subroutine would require revision to be usable for production. Sampling every other or every third cell would be possible avenues for reducing computation time, but the present version was preferred for a more accurate estimate of confidence region area. A more sophisticated method for reducing computation time would be to use a modified Golden Section Search to find the confidence region boundary outward from the center of the region, store coordinate pairs of the boundary in an m by n array, and then print the boundary on a scaled grid.

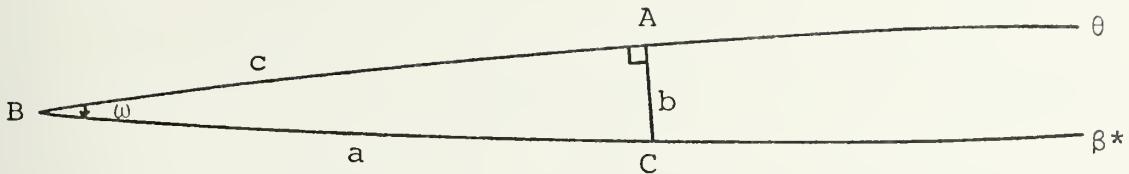
V. INTERCEPT GEOMETRY

The Fix Program produces several statistics that are useful in describing intercept geometry. The first group of these statistics is related to signal azimuths and is used to plot position lines as linear approximations of the great circle arcs in the region of the solution. In the second group are some values to which location error is sensitive.

A. PLOTTING POSITION LINES IN SOLUTION REGION

Following the optional summary of search progress, the least squares solution and the value of f^* , the computer output has a table of values in which each row corresponds to one of the intercepting DF sites. Row entries include the site identification number, the site name, the azimuth of the solution, the signal azimuth, their difference, the range in nautical miles from the solution to the site, and three values used to plot the position line relative to the least squares solution. These values are calculated by analogy with a right spherical triangle illustrated in Figure 10.

Figure 10 is labelled to correspond to the base of the polar triangle of Figure 3 with the DF site at vertex B and the least squares solution at Vertex C. Vertex angle A is a right angle formed by connecting C to the closest point of approach (CPA) of the great circle arc related to the signal



Right Spherical Triangle

FIGURE 10

azimuth θ . Arc a , connecting the DF site and least squares solution, is related to the solution azimuth, denoted β^* , and is given by equation (2-11), using the polar triangle of Figure 3. By the law of sines for spherical triangles,

$$\sin b = \sin \omega \sin a \quad (5-1)$$

The curvature of earth distance covered by arc b is closely approximated by multiplying arc b in radians by the mean earth radius of 3439.9 nautical miles, yielding the nautical mile departure of the CPA from the solution.

The site azimuth with respect to the solution, denoted γ^* , is found after calculating angle C of Figure 3 in the same manner that vertex angle B and azimuth β were calculated in equations (2-12) and (2-13). Then the azimuth of the CPA with respect to the solution is the sum of γ^* and C , in the case illustrated. Finally, the azimuth of the DF site with respect to the CPA is found by subtracting 90 degrees from the sum of γ^* and C , in the case illustrated. Similar arguments apply to three other general cases, solution east or west of the DF site and ω positive or negative, and resulting conditional equations are included in the Fix Program.

The azimuth and distance of the CPA with respect to the solution are the components of a vector which can be plotted on polar coordinate paper. A line of position perpendicular to this vector approximates the great circle arc in the region of the solution. The direction of the intercepting DF site is given by the azimuth value in the last column of the output row. The plot can also be constructed with a parallel ruler by orienting the ruler along the site azimuth with respect to the CPA, moving the free edge the distance to scale of the CPA toward the azimuth of the CPA. Such lines of position are plotted to the grid scale on several of the confidence region plots provided in the computer output section of this paper.

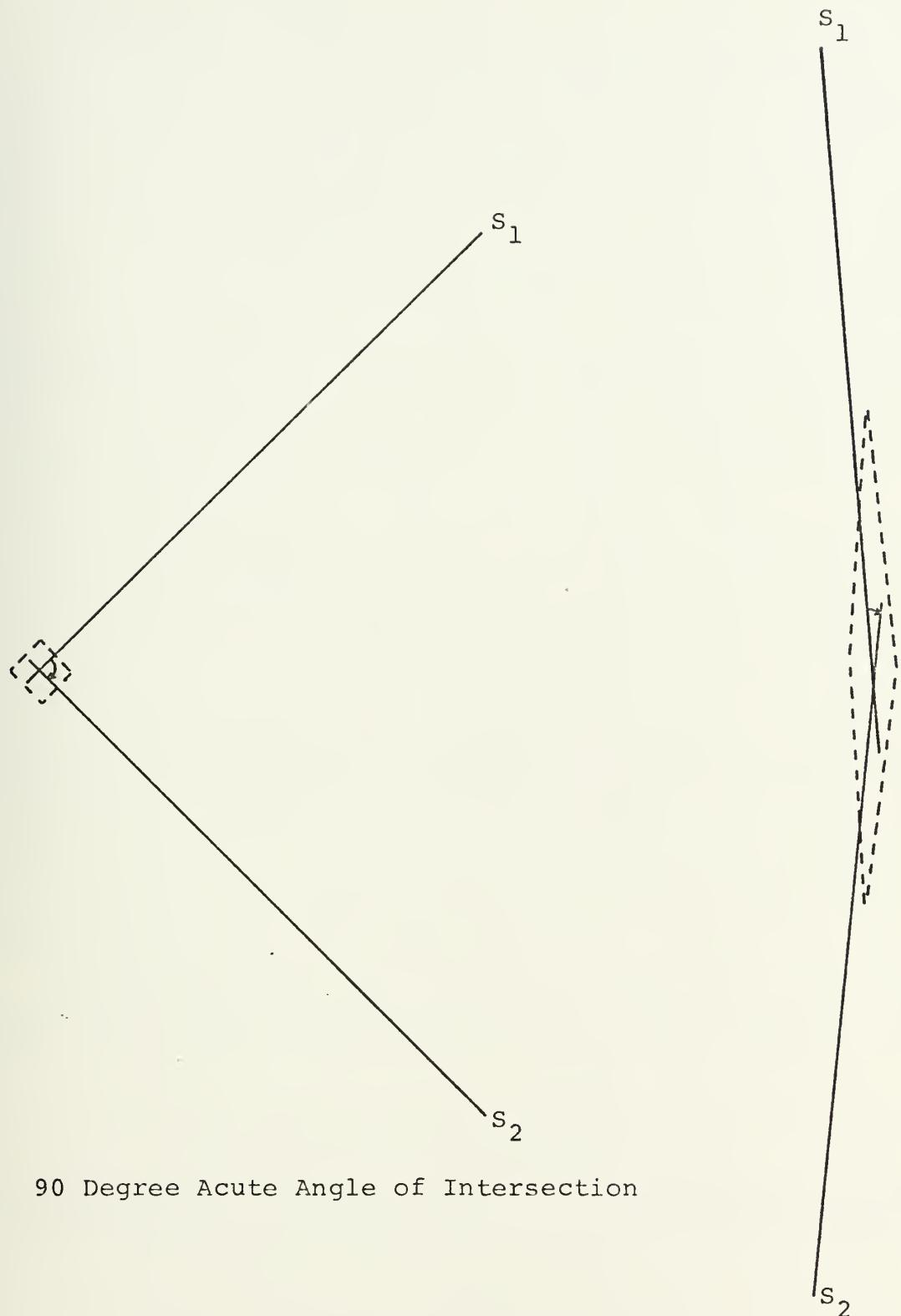
B. MEAN ACUTE ANGLE OF INTERSECTION

There are factors that may be said to reduce location error, given that signal azimuth error is present but unknown. One such factor is the number of DF sites providing signal azimuths to make a fix: the more independent measurements, the better the emitter location estimate by least squares solution. Another factor is the layout of the DF network, the obvious supposition being that a network spread over a large area produces better locations of globally distributed targets than a network confined to a small area.

A slightly less obvious statistic is the magnitude of the acute angle of intersection of great circle arcs corresponding to signal azimuths. For two DF sites participating, the dependence of location error on the acute angle of

intersection is illustrated in Figure 11. In case a, two lines of position intersect at about 90 degrees; in case b, the acute angle of intersection is about ten degrees. The difference in the sizes of regions contained by the dashed lines demonstrates the effect of two degree uncertainty of measurement on either side of a signal azimuth on the two different intersections.

Although the confidence regions described in Section IV of this paper do convey this dependence, that for case a being a near circle contained in the dashed lines and that for case b being an elongated ellipse of orientation and dimension similar to the dashed region, the acute angle of intersection statistic was found to be significant enough to warrant isolated analysis. It can be demonstrated graphically that the dependence also exists for n greater than two. For all cases, the dependence can be examined by calculating the mean value of all the acute angles of intersection formed by all possible permutations of two lines of position, given by $\binom{n}{2}$. The mean acute angle of intersection is part of the Fix Program output. Dependence of location error on this statistic is examined in later sections of this paper. For individual pairs, the acute angle of intersection is closely approximated by taking differences from the site azimuths with respect to the CPA.



a. 90 Degree Acute Angle of Intersection

b. 10 Degree Acute Angle of Intersection

FIGURE 11

C. MEAN ARC STATISTIC

A more subtle factor affecting location error is a function of arc length or distance from the DF site to the emitter location. The arc from the DF site to the emitter and all other arcs representing possible departures due to erroneous signal azimuth measurements converge at the DF site and at its antipode. Therefore, the maximum nautical mile departure of a line of position from the emitter location for constant azimuth error occurs at 90 degrees of arc, or about 5403 nautical miles, from the DF site.

The arc from the DF site to the least squares solution is given by equation (2-11). The arc statistic is equal to this arc, or to its supplement if the arc is greater than 90 degrees. The dependence of location error on the arc statistic is illustrated in Figure 12, where the solid line represents an arc from the DF site through the emitter location to the DF site antipode, and the dashed lines represent three degree departures from this arc.

The distance from the site-to-emitter arc to the departure arcs increases non-linearly from zero at the DF site

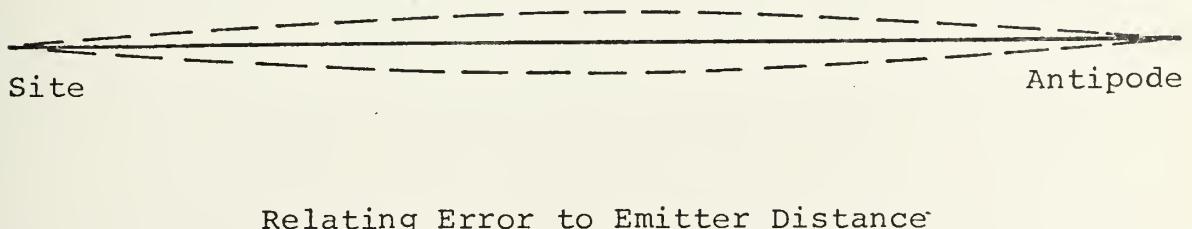


FIGURE 12

and its antipode to 180 nautical miles at the midway point. In a later section of this paper, location error sensitivity to the mean arc statistic for all DF sites participating in the location is examined.

VI. SIMULATION PROCEDURE

A modified version of the Fix Program was used to simulate operation against targets. Targets were generated by a subroutine which provided longitude and geocentric latitude describing random locations uniformly distributed over the globe. Such coordinates were saved for later comparison with the least squares solution.

The number of intercepting sites for each target was either specified for simulations on fixed degrees of freedom, or randomly generated according to a uniform distribution from three to nine. In both cases, the specific intercepting DF sites were randomly selected from a set of 19 DF sites composing the particular network being simulated. Azimuths of the target location with respect to the participating DF sites were then calculated. For each such emitter azimuth, a random error was added, with direction and magnitude of the error governed by the standard normal distribution. The emitter azimuth with error was then treated as a measured signal azimuth for input with the site identification number to the Fix Program.

During program development, the simulation procedure was used with zero signal azimuth error generated to insure that the least squares solution of the search procedure coincided with the target coordinates to at least an accuracy of three decimal places. A search termination value of one-tenth of

a nautical mile was found to be adequate, and the present combination of Golden Section Searches and second-order gradient searches was found to be satisfactory during the development simulations. The Fix Program has also been successfully tested with up to 19 participating DF sites on a single target, with a set of DF sites contained in a 180 by 240 nautical mile area, and with higher variance error distributions, including outliers.

A. GEOCENTRIC AND GEOGRAPHICAL LATITUDE

A comparison of least squares solutions using geocentric and geographical latitude in Fix Program calculations was made in order to examine the accuracy gain of the geocentric model. Errorless signal azimuths of random targets were computed using geocentric latitude in calculations. A least squares solution was then found by using geographical latitude in calculations, and the conversion of the solution to geographical latitude was omitted. This distance between the geographical coordinates of the target and the pseudo-geographical coordinates of the least squares solution measures the departure from the least squares location due to use of geographical latitude on an assumed sphere.

Radial departures for two degrees of freedom were occasionally devastating (424 nautical miles in one case). Using an imaginary network, called the national network, and degrees of freedom uniformly distributed from three to nine, the mean radial departure for 148 samples was 6.8 nautical miles. The direction of such departures tends toward the equator for targets in both northern and southern hemispheres.

VII. LOCATION ERROR ANALYSIS WITH FIXED DEGREES OF FREEDOM

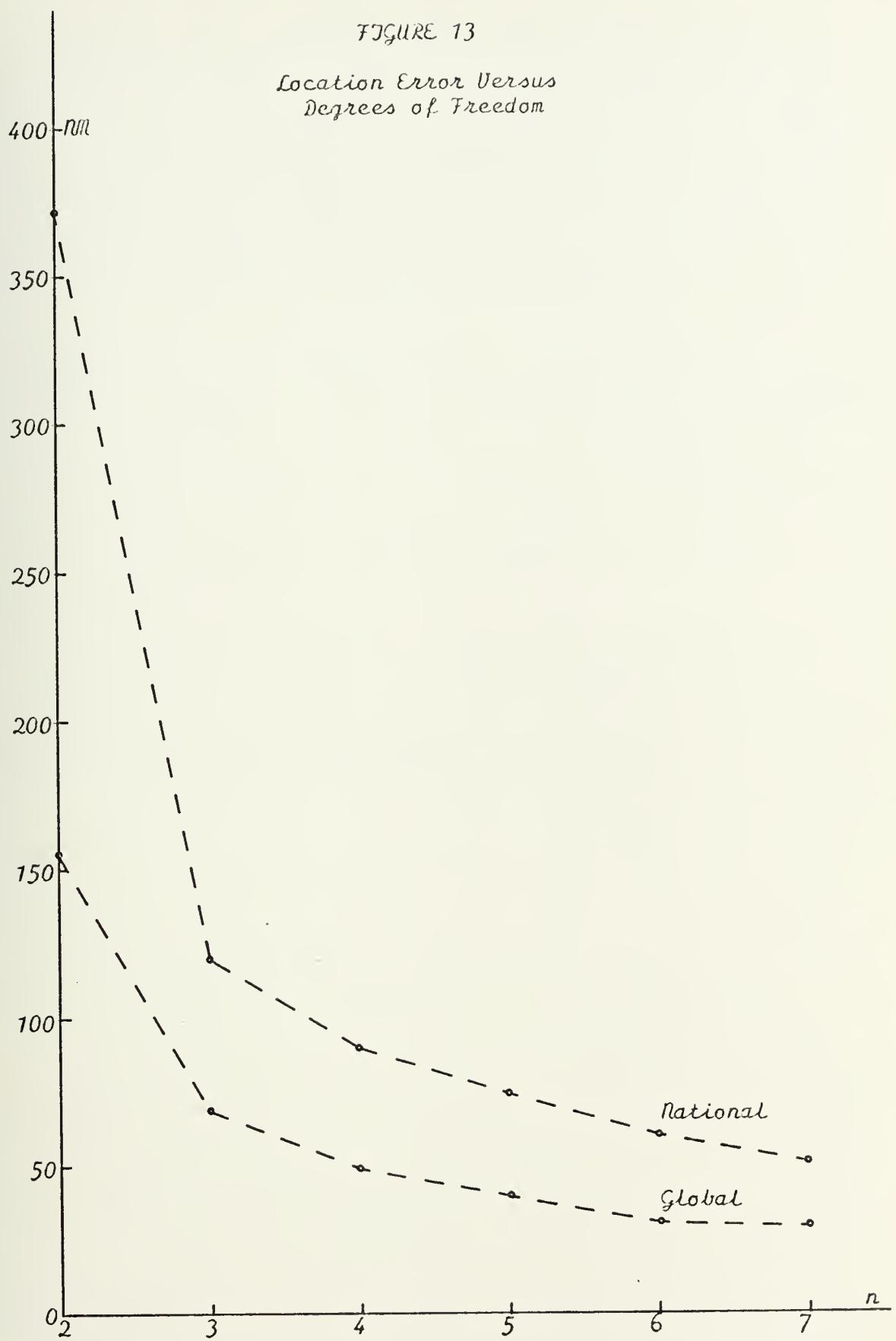
The sensitivity of location error to several factors was examined in simulations with degrees of freedom fixed from two to seven. From 100 to 200 samples were generated at each level for each of the networks. All of the following plots are drawn to the same scale to simplify between plot comparisons of the effects of different degrees of freedom and different network configurations. Values too large for the common scale are entered in a row at the upper limit of the plot above an off of scale indicator.

A. COMPARISON OF ERROR ON NATIONAL AND GLOBAL NETWORKS

Member sites of the national network varied in longitude from Attu to Boston and in latitude from Anchorage to Honolulu. The specific sites are identified in the second data statement of the Fix Program. Sites in the widely distributed global network included Anchorage, Bombay, Boston, Buenos Aires, Cape Town, Christchurch, Dakar, Honolulu, Houston, Lima, Naples, Papeete, Perth, Recife, Seattle, Seychelles, Tehran, Tokyo and Wake.

Figure 13 is a plot of mean location error from each of the sample groups against the number of intercepting sites. Dashed lines connect the means of each network. The plot shows nearly double the location error in the national network even at higher degrees of freedom.

FIGURE 13
Location Error Versus
Degrees of Freedom



With all other factors random, the average accuracy of a network is dependent upon the network layout and the variance of signal azimuth measurement error. For a given network, the mean location error would vary in the same direction as changes in such variance, in all of the following simulations held constant at one degree.

B. ERROR VERSUS MEAN ACUTE ANGLE OF INTERSECTION

Based on samples from the national network, Figures 14, 15 and 16 are scatter plots of location error versus the mean acute angle of intersection explained in section V-B. Degrees of freedom were fixed at three, five and seven, respectively.

The strong correlation in the national network can be interpreted as a reduction of both the mean and variance of location error as intersections of great circle arcs are less acute. Rough scatter plots for two, four and six degrees of freedom showed the same correlation, naturally most pronounced at two degrees of freedom. The set of plots shows a shrinking of the range of the mean acute angle of intersection statistics as degrees of freedom are increased.

Figures 17, 18 and 19 are similar scatter plots with entries from simulations on the global network. The correlation is still evident at three degrees of freedom but becomes masked at higher levels as the layout of the network reduces the chance of producing a combination of target and intercepting sites with a small mean acute angle of inter-

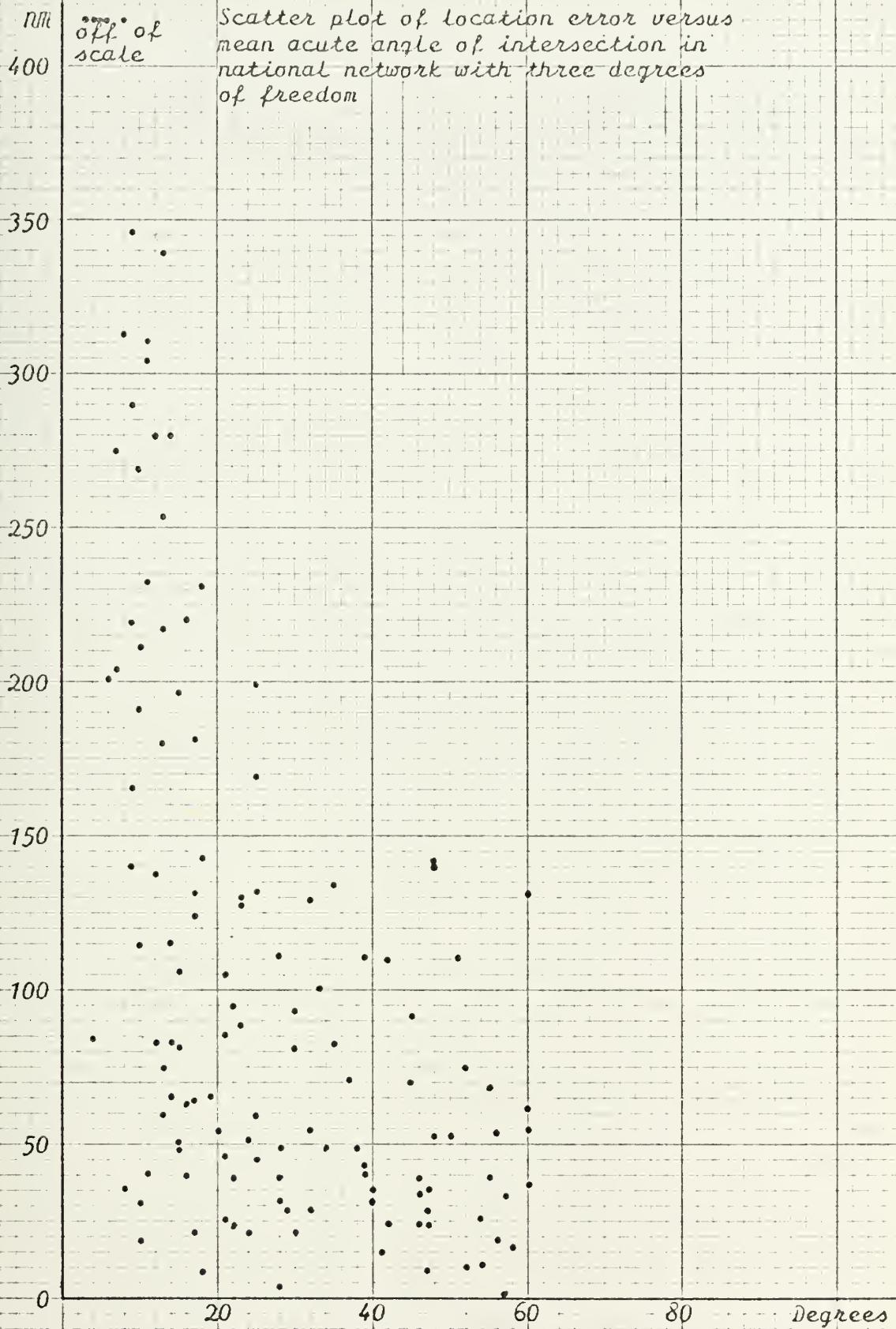


FIGURE 14

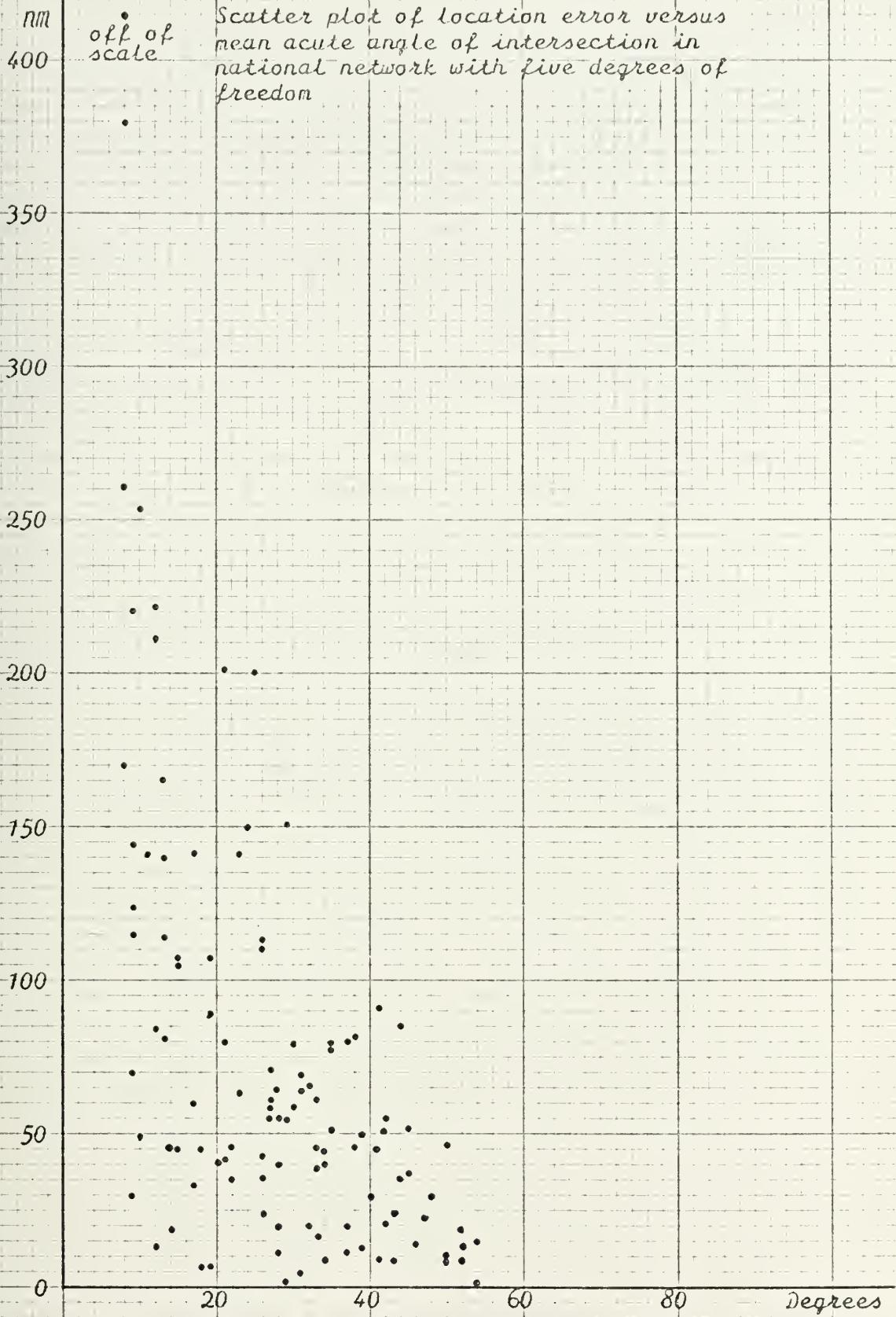
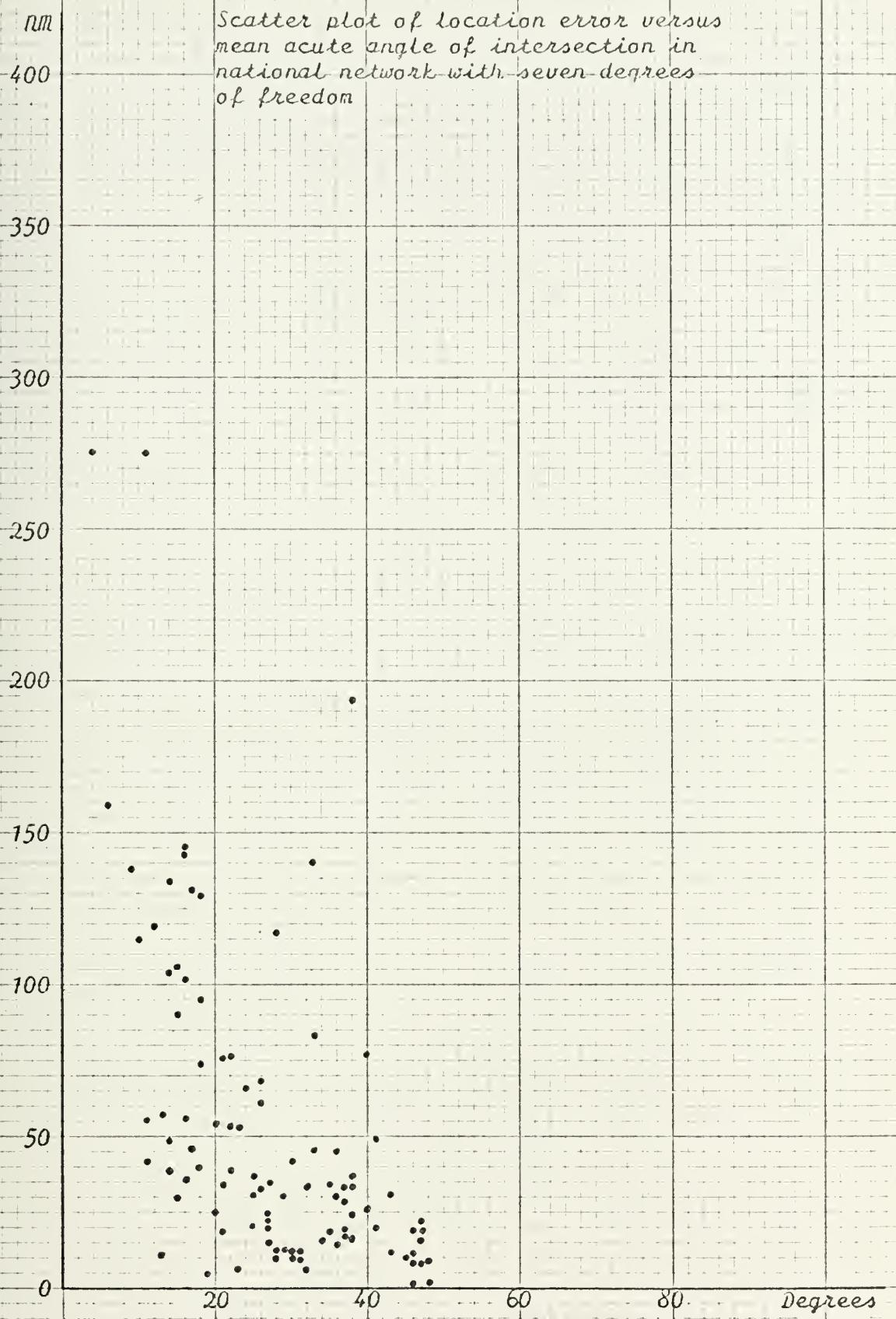


FIGURE 15

Scatter plot of location error versus
mean acute angle of intersection in
national network with seven-degrees
of freedom



NM

400

350

300

250

200

150

100

50

0

Scatter plot of location error versus mean acute angle of intersection in global network with three degrees of freedom

20 40 60 80 Degrees

FIGURE 17

nm
400

350

300

250

200

150

100

50

0

Scatter plot of location error versus mean acute angle of intersection in global network with five degrees of freedom

20

40

60

80

Degrees

FIGURE 18

Scatter plot of location error versus
mean acute angle of intersection in
global network with seven degrees of
freedom

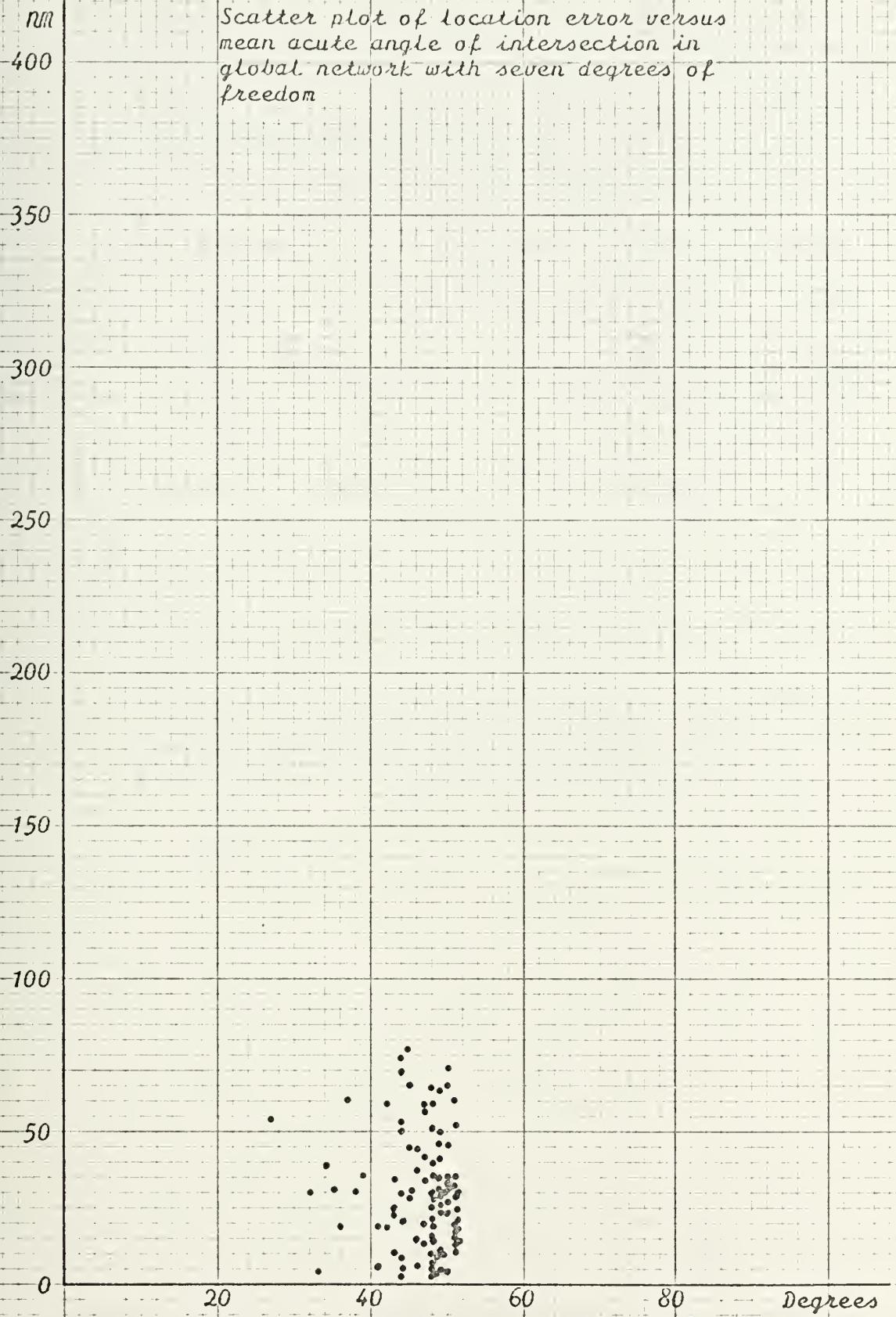


FIGURE 19

section. Consequently, shrinking of the range of the statistic is extremely evident on Figures 18 and 19.

C. ERROR VERSUS MEAN ARC STATISTIC

Figures 20, 21 and 22 are scatter plots of national network location error versus the mean arc statistic (explained in section V.C) for three, five and seven degrees of freedom, respectively. At all levels, the large location errors tend to be associated with the high end of the range of mean arc statistics. The strength of the correlation appears to increase as degrees of freedom increase on successive plots.

Figures 23, 24 and 25 are similar plots for the global network. Again, the correlation is masked after three degrees of freedom with the shrinkage effect extremely pronounced as the degrees of freedom increase.

nm
400

Scatter plot of location error
versus mean arc statistic in
national network with three
degrees of freedom

off of scale

350

300

250

200

150

100

50

0

20 40

60

80

Degrees

FIGURE 20

400
350
300
250
200
150
100
50
0

Scatter plot of location error
versus mean arc statistic in
national network with five
degrees of freedom

off of
scale

20 40 60 80 Degrees

FIGURE 21

400
350
300
250
200
150
100
50
0

Scatter plot of location error
versus mean arc statistic in
national network with seven
degrees of freedom

20 40 60 80

Degrees

FIGURE 22

RM

400

350

300

250

200

150

100

50

0

Scatter plot of location error
versus mean arc statistic in
global network with three
degrees of freedom

off of scale

20

40

60

80

Degrees

FIGURE 23

Nil
400

Scatter plot of location error
versus mean arc statistic in
global network with five
degrees of freedom

350

300

250

200

150

100

50

0

20

40

60

80

Degrees

FIGURE 24

Scatter plot of location error
versus mean arc statistic in
global network with seven
degrees of freedom

350

300

250

200

150

100

50

0

20

40

60

80

Degrees

FIGURE 25

IIX. LOCATION ERROR ANALYSIS WITH VARYING DEGREES OF FREEDOM

Simulations were conducted on both networks with degrees of freedom generated uniformly from three to nine. Chi-square contour confidence regions were generated for each target's least squares solution. Using a confidence level of 90 percent, national network confidence regions contained 90.38 percent of the generated targets. Results for the global network were 90.44 percent. Such close agreement between the confidence level and containment in the simulations is dependent upon a reliable standard normal azimuth error generator. In production use, such reliability could only be achieved by having a good estimate of DF site variances.

A. NATIONAL NETWORK

Figure 26 is a scatter plot of confidence region area versus location error. Again using a common scale to aid network comparison, eleven points off the scale are entered above or to the right of an indicator. Although there is one hopeful cluster near the plot origin, the large variance in location errors at given confidence region areas is indicative of the futility of predicting location error based on the size of this reliable confidence region model.

Figure 27 is a multi-level histogram of location error at the various degrees of freedom, demonstrating the same sort of correlation evident in Figure 13. Figure 28

nm^2
80000

70000

60000

50000

40000

30000

20000

10000

0

off of scale

off
of
scale

Scatter plot of confidence
region area versus location
error in national network

50

100

150

200

250 nm

FIGURE 26

run
400

350

300

250

200

150

100

50

0

Location error versus
degrees of freedom in
national network

df
FIGURE 27

RM

400

350

300

250

200

150

100

50

0

Scatter plot of location error
versus mean acute angle of
intersection in national network

20 40 60 80 degrees

FIGURE 28

illustrates the correlation between location error and the mean acute angle of intersection, present even with mixed degrees of freedom. The same holds true in Figure 29, a plot of location error versus the mean arc statistic.

B. GLOBAL NETWORK

Figures 30 through 33 are similar plots for the global network. On Figure 30, location error and confidence region area appear independent. Figure 31 shows good correlation between location error and degrees of freedom. Figures 32 and 33 show nearly complete masking of the previously established correlations due to the layout of the network.

nm
400

350

300

250

200

150

100

50

0

Scatter plot of location error
versus mean arc statistic in
national network.

20 40 60 80 Degrees

FIGURE 29

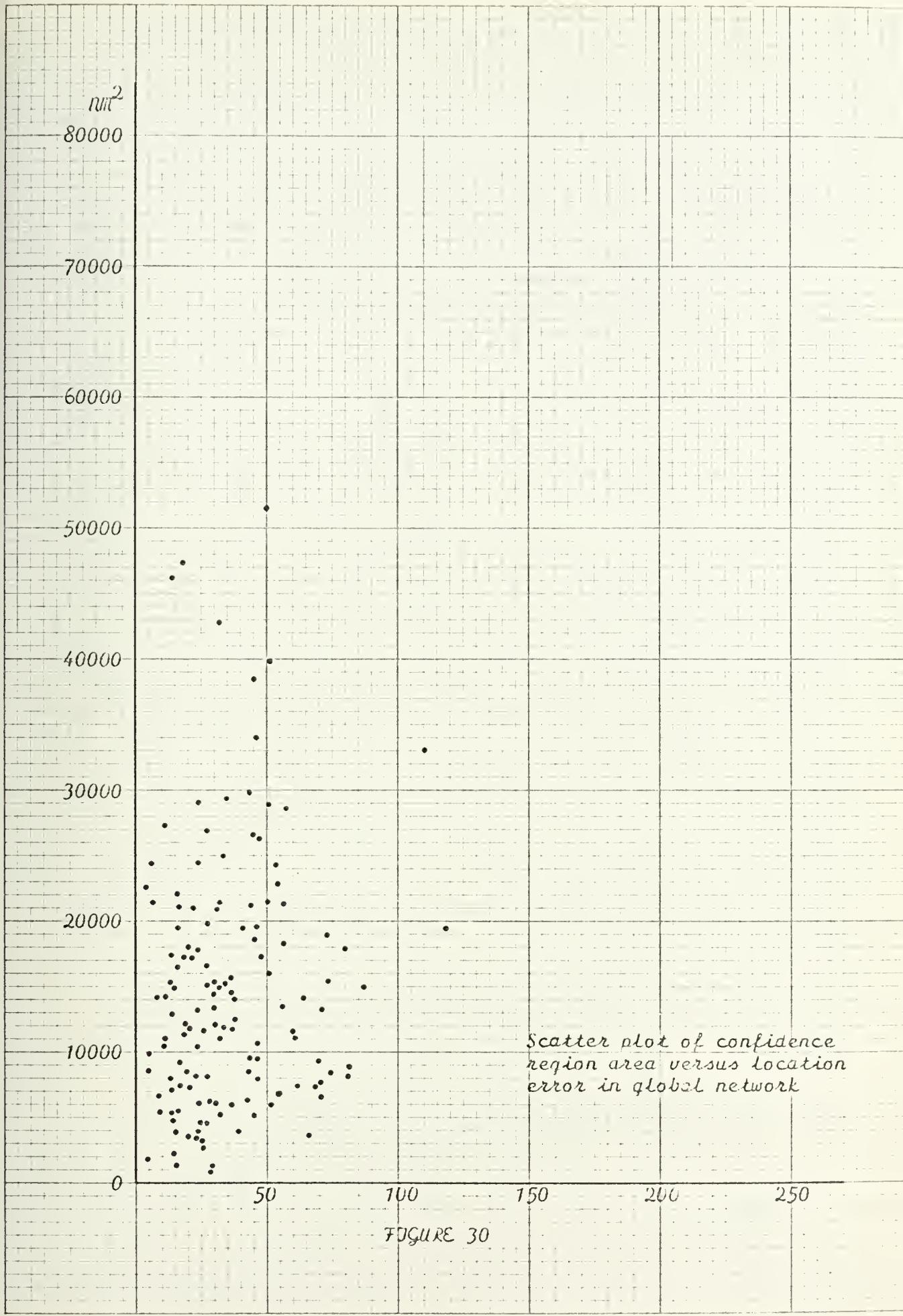


FIGURE 30

NN

400

350

300

250

200

150

100

50

0

Location error versus
degrees of freedom in
global network

3 4 5 6 7 8 9

df

FIGURE 31

nm

400

350

300

250

200

150

100

50

0

Scatter plot of location error
versus mean acute angle of
intersection in global network

FIGURE 32

100

350

300

250

200

150

100

50

0

scatter plot of location error
versus mean arc statistic in
global network

20

40

60

80

Degrees

FIGURE 33

IX. CONCLUSION

The Fix Program and Chi-Square contour confidence region subroutine could be integrated into the software of position location systems by direction finding. The present program (with target and azimuth error generators included) could be used to simulate an existing DF network for accuracy studies on potential targets in different regions of the globe or for studies on optimal placement of DF sites in a proposed network.

The relationships demonstrated in the sections on location error analysis indicate that larger location errors accompany contraction of the network layout; however, this factor would be mitigated by a practice of assigning regional rather than globally distributed targets to the network for prosecution.

Countering the widespread belief that small confidence regions indicate accurate locations and large confidence regions indicate poor locations, it has been demonstrated that for a reliable model location error and confidence region area are nearly independent.

The relationships examined as parts of location error analysis suggest an alternative answer for the question of how accurate is a given location. For an existing network, data could be collected from locations of known transmitters. Such data would include statistics on number of signal

azimuths available to make the fix, mean acute angle of intersection and mean arc statistic. With known location error as the dependent variable, multiple linear regression analysis could be used to develop regression and correlation coefficients for predicting location error as a function of independent variables measured or calculated at the time of location.

The workability of such an approach was not demonstrated in the present paper for several reasons: (1) The effort required to perform the multiple linear regression (using a standard program from the IBM scientific subroutine package) would be a project in itself and would be best reserved for actual data from an existing network. (2) System operators would be aware of other measurable factors for inclusion as independent variables in the regression analysis, such as variance on given signal azimuth measurements, signal strength and measurement duration.

FIX PROGRAM

```

REAL LAMBDA,N,NL,NP,NLL,NPP,NPL
DIMENSION DFSITE(6,19),DOMEGA(19),SYMBOL(4),BETA(19),
1STAATZ(19)
COMMON/SITES/SITE(10,19),NUMBER
COMMON/SIGNAL/THETA(19)
DATA TWOPI/6.283185/,PI/3.141593/,RADDEG/57.29578/,
1DEGRAD/1.745329E-2/,FNCECC/1.006768/,SCREEN/.999999999,
2PIHALF/1.570796/,SYMBCL/'N','S','E','W'/
DATA DFSITE/
C IDENT LONG LAT SITE NAME
1 1., 210.117, 61.217, 'ANCH', 'ORAG', 'E', ','
2 2., 173.250, 52.933, 'ATTU', 'ON', ','
3 3., 288.933, 42.350, 'BOST', 'ON', ','
4 4., 262.600, 27.800, 'CORP', 'US C', 'HRIS', ','
5 5., 254.983, 39.717, 'DENV', 'ER', ','
6 6., 276.950, 42.333, 'DETROIT', 'OIT', ','
7 7., 253.517, 31.750, 'EL PASO', 'ASO', ','
8 8., 247.983, 46.600, 'HELENA', 'NA', ','
9 9., 202.133, 21.317, 'HONO', 'LULU', ','
1 10., 241.750, 34.050, 'LOS ANGELES', 'LES', ','
2 11., 279.800, 25.767, 'MIAMI', 'I', ','
3 12., 182.633, 28.217, 'MIDWAY', 'AY', ','
4 13., 266.783, 44.983, 'MINN', 'EAPO', 'LIS', ','
5 14., 273.200, 36.150, 'NASH', 'VILL', 'E', ','
6 15., 269.883, 29.967, 'NEW ORLEANS', 'ANS', ','
7 16., 283.767, 38.667, 'NORFOLK', 'OLK', ','
8 17., 240.200, 39.517, 'RENO', 'T', ','
9 18., 237.667, 47.600, 'SEATTLE', 'TLE', ','
1 19., 262.667, 37.683, 'WICHITA', 'ITA', ','
/READ (5,1) KILL,CUT
0900 READ (5,1) NUMBER
IF (NUMBER.EQ.0) GO TO 4200
WRITE (6,9)
9 FORMAT ('1')
1 FORMAT (I2,F8.3)
DO 1000 K = 1,NUMBER
READ (5,1) ID,THETA(K)
THETA(K)=THETA(K)*DEGRAD
SITE(1,K)=DFBSITE(2,ID)*DEGRAD
SITE(2,K)=SIN(SITE(1,K))
SITE(3,K)=COS(SITE(1,K))
SITE(4,K)=ATAN(0.9932773*TAN(DFBSITE(3,ID)*DEGRAD))
SITE(5,K)=SIN(SITE(4,K))
SITE(6,K)=COS(SITE(4,K))
SITE(7,K)=DFBSITE(1,ID)
SITE(8,K)=DFBSITE(4,ID)
SITE(9,K)=DFBSITE(5,ID)
1000 SITE(10,K)=DFBSITE(6,ID)
C PERFORM ALTERNATING GOLDEN SECTION SEARCHES ON LONGITUDE
C AND LATITUDE
PSI=0.
LAMBDA=0.
SAVEL=LAMBDA
SAVEP=PSI
DEGF=615600.
CO 1400 J = 1,5
DIV=J
X=1./DIV*TWOPI
Y=0.
CALL SEARCH(LAMBDA,PSI,X,Y,DEGF)
IF (KILL.EQ.2) GO TC 1100
CALL SHIFT(SAVEL,SAVEP,LAMBDA,PSI,1,DEGF,DIST)
1100 CONTINUE
X=0.
Y=1./DIV*PI
CALL SEARCH(LAMBDA,PSI,X,Y,DEGF)
IF (KILL.EQ.2) GO TC 1200
CALL SHIFT(SAVEL,SAVEP,LAMBDA,PSI,1,DEGF,DIST)
GO TO 1300
1200 CALL AZDIST(SAVEL,SAVEP,LAMBDA,PSI,AZGO,DIST)

```



```

SAVEL=LAMBDA
SAVEP=PSI
1300 IF (DIST.LE.500.) GO TO 1500
1400 CONTINUE
1500 M=0
1600 M=M+1
COSLAM=CCS(LAMBDA)
SINLAM=SIN(LAMBDA)
COSPSI=COS(PSI)
SINPSI=SIN(PSI)
FL=0.
FP=0.
FLL=0.
FPP=0.
FPL=0.
C EVALUATE FIRST AND SECOND PARTIALS OF OBJECTIVE FUNCTION
DO 2450 K = 1, NUMBER
A=1.
C=0.
UA=SINPSI*SITE(6,K)
UB=COSPSI*SITE(5,K)
UC=COSLAM*SITE(3,K)+SINLAM*SITE(2,K)
UD=-SINLAM*SITE(3,K)+COSLAM*SITE(2,K)
UE=COSPSI*SITE(6,K)
UF=SINPSI*SITE(5,K)
UG=UF+UE*UC
UH=UB-UA*UC
UI=-UE*UD
N=UA-UB*UC
NL=-UB*UD
NP=UE+UF*UC
NLL=UB*UC
NPP=-N
NPL=UF*UD
IF (UG.GE.1.) UG=SCREEN
IF (UG.LE.-1.) UG=-SCREEN
D=SIN(ARCOS(UG))
UJ=1./D
DL=UJ*UG*UI
DP=-UJ*UG*UH
DLL=UJ*(UG*UE*UC+UI*UI-DL*DL)
DPP=UJ*(UG*UG-UH*UH-DP*DP)
DPL=UJ*(UG*UA*UD+UH*UI-DP*DL)
COSB=N*UJ
IF (COSB.GE.1.) COSB=SCREEN
IF (COSB.LE.-1.) COSB=-SCREEN
B=ARCOS(COSB)
COTB=COTAN(B)
CSCB=1./SIN(B)
BL=UJ*(DL*COTB-NL*CSCB)
BP=UJ*(DP*COTB-NP*CSCB)
UK=UJ*CSCB
BLL=UK*(UJ*DL*(NL-DL*COSB)+DLL*COSB-BL*DL*CSCB-NLL+
1BL*NL*COTB)
BPP=UK*(UJ*DP*(NP-DP*COSB)+DPP*COSB-BP*DP*CSCB-NPP+
1BP*NP*COTB)
BPL=UK*(UJ*DP*(NL-DL*COSB)+DPL*COSB-BP*DL*CSCB-NPL+
1BP*NL*COTB)
TEST=SITF(1,K)-LAMBDA
IF (TEST) 1700,2000,1800
1700 IF (TEST+PI) 1900,2000,2000
1800 IF (TEST-PI) 1900,2000,2000
1900 A=-1.
C=TWOP
2000 TEST=A*B+C-THETA(K)
BETA(K)=(A*B+C)*RADDEG
IF (ABS(TEST)-PI) 2400,2400,2100
2100 IF (TEST) 2200,2400,2300
2200 C=C+TWOP
GO TO 2400
2300 C=C-TWOP
2400 CMEGA=A*B+C-THETA(K)

```



```

DCMEGA(K)=OMEGA*RADDEG
UL=2.*A*(A*B+C-THETA(K))
FL=FL+UL*BL
FP=FP+UL*BPP
FLL=FLL+2.*BL*BL+UL*BLL
FPP=FPP+2.*BP*BP+UL*BPP
FPL=FPL+2.*BP*BL+UL*BPL
2450 CONTINUE
UM=1./(FLL*FPP-FPL*FPL)
X=(FPL*FP-FPP*FL)*UM
Y=(FPL*FL-FLL*FP)*UM
C IF EITHER X OR Y ARE GREATER THAN PI AND UNTIL WITHIN 30
C NM OF THE INDICATED MOVE, CONVERT (X,Y) TO A RADIAN UNIT
C VECTOR
IF (AMAX1(X,Y).GE.PI) GO TO 2500
GOX=LAMBDA*X
GOY=PSI+Y
CALL AZDIST(LAMBDA,PSI,GOX,GOY,AZGO,DIST)
IF (DIST.LE.30.) GO TO 2600
2500 CONTINUE
DIST=SQRT(X*X+Y*Y)
X=X/DIST
Y=Y/DIST
2600 CONTINUE
C REFINE INDICATED MOVE BY A GOLDEN SECTION SEARCH
CALL SEARCH(LAMBDA,PSI,X,Y,DEGF)
IF (KILL.EQ.2) GO TO 2700
CALL SHIFT(SAVEL,SAVEP,LAMBDA,PSI,2,DEGF,DIST)
GO TO 2800
2700 CALL AZDIST(SAVEL,SAVEP,LAMBDA,PSI,AZGO,DIST)
SAVEL=LAMBDA
SAVEP=PSI
C CHECK FOR MOVE LESS THAN MINIMUM STEP
2800 IF (DIST.LE.CUT) GO TO 3300
C SEARCHING STOPS AFTER 30 ITERATIONS UNLESS SEARCH PROGRESS
C IS REQUESTED
IF (KILL.EQ.2) GO TO 2900
IF (M.EQ.30) GO TO 3300
2900 CONTINUE
C IF THE SOLUTION HAS NOT BEEN FOUND, APPLY THE GOLDEN
C SECTION SEARCH AFTER EVERY TENTH ITERATION.
IF (MOD(M,10).NE.0) GO TO 1600
X=PIHALF
Y=0.
CALL SEARCH(LAMBDA,PSI,X,Y,DEGF)
IF (KILL.EQ.2) GO TO 3000
CALL SHIFT(SAVEL,SAVEP,LAMBDA,PSI,1,DEGF,DIST)
GO TO 3100
3000 SAVEL=LAMBDA
SAVEP=PSI
3100 X=0.
Y=PIHALF
CALL SEARCH(LAMBDA,PSI,X,Y,DEGF)
IF (KILL.EQ.2) GO TO 3200
CALL SHIFT(SAVEL,SAVEP,LAMBDA,PSI,1,DEGF,DIST)
GO TO 1600
3200 SAVEL=LAMBDA
SAVEP=PSI
GO TO 1600
3300 DEGLAM=LAMBDA*RADDEG
IF (DEGLAM.GT.180.) GO TO 3400
X=SYMBOL(3)
GO TO 3500
3400 DEGLAM=360.-DEGLAM
X=SYMBOL(4)
3500 NDEGX=DEGLAM
FDEG=NDEGX
NMINX=(DEGLAM-FDEG)*60.+5
DEGPHI=ATAN(FNCECC*TAN(PSI))*RADDEG
IF (DEGPHI.LT.0.) GO TO 3600
Y=SYMBOL(1)
GO TO 3700

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3600 Y=SYMBOL(2)
DEGPHI=-DEGPHI
3700 NDEGY=DEGPHI
FDEG=NDEGY
NMINY=(DEGPHI-FDEG)*60.+.5
WRITE (6,2) NDEGX,NMINX,X,NDEGY,NMINY,Y
2 FORMAT ('0',30X,I3,'-',I2,A1,',',I3,'-',I2,A1,' IS THE
1' LEAST SQUARES SOLUTION.')
WRITE (6,3) DEGF
3 FORMAT (31X,' OBJECTIVE FUNCTION =',F8.2,' DEGREES',
1' SQUARED.')
WRITE (6,4) NUMBER
4 FORMAT (30X,I3,' DEGREES OF FREEDOM')
WRITE (6,5)
5 FORMAT ('0',45X,'SOLUTION SIGNAL SITE',/,30X,
1' ID INTERCEPT AZIMUTH AZIMUTH RANGE PLOT',
2' THETA VALS',/,30X,' NR SITE NAME BETA - THETA',
3'= OMEGA NM NORM NM AZ')
SUMARC=0.
DC 4000 K = 1, NUMBER
ID=SITE(7,K)+.5
DTTHETA=THETA(K)*RADDEG
CALL AZDIST(LAMBDA,PSI,SITE(1,K),SITE(4,K),SITEAZ,
1DTST)
NMSITE=DIST+.5
ARCC=(DIST/3439.92)
ANGLEB=ABS(DOMEGA(K))*DEGRAD
ARCB=ARSIN(SIN(ANGLEB)*SIN(ARCC))
ANGLEA=ARCCOS(TAN(ARCB)*COTAN(ARCC))
STAT=ARCC*RADDEG
IF (STAT.GT.90.) STAT=180.-STAT
IF (DOMEGA(K).GT.0.) GO TO 3800
AZTHET=SITEAZ-ANGLEA*RADDEG
IF (AZTHET.LT.0.) AZTHET=AZTHET+360.
STAАЗ{K}=AZTHET+90.
IF (STAАЗ{K}.GT.360.) STAАЗ{K}=STAАЗ{K}-360.
GO TO 3900
3800 AZTHET=SITEAZ+ANGLEA*RADDEG
IF (AZTHET.GT.360.) AZTHET=AZTHET-360.
STAАЗ{K}=AZTHET-90.
IF (STAАЗ{K}.LT.0.) STAАЗ{K}=STAАЗ{K}+360.
3900 NMTHET=ARCB*3439.92
WRITE (6,6) ID,SITE(8,K),SITE(9,K),SITE(10,K),BETA(K),
1DTTHETA,DOMEGA(K),NMSITE,AZTHET,NMTHET,STAАЗ{K}
6 4000 FORMAT (30X,I3,1X,3A4,F8.3,F8.3,F7.3,I6,F6.1,I4,F6.1)
SUMARC=SUMARC+STAT
DNUM=NUMBER
AVGARC=SUMARC/DNUM
WRITE (6,7) AVGARC
7 FORMAT ('0',30X,'MEAN ARC STATISTIC =',F5.1,' DEGREES.
CCUNT=0.
SUMANG=0.
NEND=NUMBER-1
DO 4100 I = 1,NEND
JBEGIN=I+1
DO 4100 J = JBEGIN,NUMBER
ACUTE=ABS(STAAZ(I)-STAAZ(J))
IF (ACUTE.GT.180.) ACUTE=ACUTE-180.
IF (ACUTE.GT.90.) ACUTE=180.-ACUTE
SUMANG=SUMANG+ACUTE
4100 COUNT=COUNT+1.
AVGANG=SUMANG/COUNT
WRITE (6,8) AVGANG
8 FORMAT (31X,'MEAN ACUTE ANGLE OF INTERSECTION =',F5.1,
1' DEGREES.')
CALL PLOT (LAMBDA,PSI,DEGF,NUMBER)
GO TO 0900
4200 CONTINUE
STOP
END

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SUBROUTINE SEARCH(L,P,X,Y,SAVEF)
C GOLDEN SECTION SEARCH
REAL LAMBDA(2),L
DIMENSION DEL(2),FDEL(2),PSI(2)
CCMON/SITES/SITE(10,19),NUMBER
CCMON/SIGNAL/THETA(19)
DATA TWOPI/6.283185/,PIHALF/1.570796/,PI/3.141593/,
1 RADDEG/57.29578/
SAVEL=L
SAVEP=P
IFLAG=0
BEGIN=-1.
END=1.
J=1
1000 DEL(J)=0.618*BEGIN+0.382*END
GO TO 1200
1100 IFLAG=2
DEL(J)=0.382*BEGIN+0.618*END
1200 FDEL(J)=0.
LAMBDA(J)=L+X*DEL(J)
PSI(J)=P+Y*DEL(J)
C FIND CORRESPONDING TEST LONGITUDE IN THE INTERVAL
C (0,TWOPI).
IF (LAMBDA(J).GT.TWOPI) LAMBDA(J)=LAMBDA(J)-TWOPI
IF (LAMBDA(J).LT.0.) LAMBDA(J)=TWOPI+LAMBDA(J)
C FIND CORRESPONDING TEST LATITUDE IN THE INTERVAL
C (-PIHALF,PIHALF).
IF (PSI(J).GT.PIHALF) PSI(J)=PI-PSI(J)
IF (PSI(J).LT.-PIHALF) PSI(J)=-PI-PSI(J)
COSLAM=COS(LAMBDA(J))
SINLAM=SIN(LAMBDA(J))
COSPSI=COS(PSI(J))
SINPSI=SIN(PSI(J))
VARLAM=LAMBDA(J)
DO 1700 K = 1,NUMBER
CALL AZCCMP(COSLAM,SINLAM,COSPSI,SINPSI,K,VARLAM,
1 A,B,C)
TEST=A*B+C-THETA(K)
IF (ABS(TEST)-PI) 1600,1600,1300
1300 IF (TEST) 1400,1600,1500
1400 C=C+TWOPI
GO TO 1600
1500 C=C-TWOPI
1600 CMEGA=A*B+C-THETA(K)
1700 FDEL(J)=FDEL(J)+CMEGA**2
IF (IFLAG.GT.1) GO TO 1800
J=2
GO TO 1100
1800 CONTINUE
C RETAIN VALUES CORRESPONDING TO CURRENT MINIMUM, ADJUST
C INTERVAL AND TEST NEXT DELTA SYMMETRIC WITH CURRENT DELTA.
IF (FDEL(1)-FDEL(2)) 1900,1900,2000
1900 END=DEL(2)
PSI(2)=PSI(1)
LAMBDA(2)=LAMBDA(1)
FDEL(2)=FDEL(1)
DEL(2)=DEL(1)
J=1
GO TO 2100
2000 BEGIN=DEL(1)
PSI(1)=PSI(2)
LAMBDA(1)=LAMBDA(2)
FDEL(1)=FDEL(2)
DEL(1)=DEL(2)
J=2
C TEST SIZE OF THE INTERVAL
2100 IF ((END-BEGIN).LE.0.01) GO TO 2200
GO TO (1000,1100), J

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```

2200  MIN=1
      IF (J.EQ.1) MIN=2
      L=LAMBDA(MIN)
      P=PSI(MIN)
      DELTA=DEL(MIN)
      DEGF=FDEL(MIN)*RADDEG**2
      IF (DEGF.LE.SAVEF) GO TO 2300
      L=SAVEL
      P=SAVEP
      RETURN
2300  SAVEF=DEGF
      RETURN
      END

      SUBROUTINE AZDIST(RLAM,RPSI,TLAM,TPSI,AZ,DIST)
C AZDIST COMPUTES AZIMUTH AND DISTANCE OF A TARGET WITH
C RESPECT TO A REFERENCE POINT
      DOUBLE PRECISION COSRP,SINRP,COSTP,SINTP,UC,RNUM,UG,
1 DEN,COSB,REFLAM,REFPSI,TARLAM,TARPSI,ARC
      DATA TWOPI/6.283185/,RADDEG/57.29578/,PI/3.141593/
      REFLAM=RLAM
      REFPSI=RPSI
      TARLAM=TLAM
      TARPSI=TPSI
      COSRP=DCOS(REFPSI)
      SINRP=DSIN(REFPSI)
      COSTP=DCOS(TARPSI)
      SINTP=DSIN(TARPSI)
      UC=DCOS(TARLAM)*DCOS(REFLAM)+DSIN(TARLAM)*DSIN(REFLAM)
      RNUM=SINTP*COSRP-COSTP*SINRP*UC
      UG=SINTP*SINRP+COSTP*COSRP*UC
      IF (UG.GE.1.0D0) UG=1.0D0
      IF (UG.LE.-1.0D0) UG=-1.0D0
      ARC=CARCCS(UG)
      DEN=DSIN(ARC)
      IF (DABS(ARC).LT.0.1D-4) GO TO 1600
      IF (DABS(RNUM).GT.0.0D-10) GO TO 1000
      COSB=0.0D0
      GO TO 1100
1000  COSB=RNUM/DEN
1100  CONTINUE
      CCSB=RNUM/DEN
      IF (COSB.GE.1.0D0) COSB=1.0D0
      IF (COSB.LE.-1.0D0) COSB=-1.0D0
      BETA=DARCOS(COSB)
      TEST=REFLAM-TARLAM
      IF (TEST) 1200,1500,1300
1200  IF (TEST+PI) 1400,1500,1500
1300  IF (TEST-PI) 1400,1500,1500
1400  BETA=TWOPI-BETA
1500  AZ=BETA*RADDEG
      DIST=ARC*3439.92D0
      GO TO 1700
1600  AZ=0.
      DIST=0.
1700  CONTINUE
      RETURN
      END

      SUBROUTINE AZCOMP(CCSLAM,SINLAM,COSPSI,SINPSI,K,
C AZCOMP COMPUTES POINT AZIMUTHS WITH RESPECT TO SITES
1 LAMBDA,A,B,C)
      REAL N,LAMBDA
      COMMON/SITES/SITE(10,19),NUMBER
      DATA PI/3.141593/,TWOPI/6.283185/,SCREEN/.9999999/
      A=1.
      C=0.
      UC=COSLAM*SITE(3,K)+SINLAM*SITE(2,K)
      N=SINPSI*SITE(6,K)-COSPSI*SITE(5,K)*UC
      UG=SINPSI*SITE(5,K)+COSPSI*SITE(6,K)*UC

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IF (UG.GE.1.) UG=SCREEN
IF (UG.LE.-1.) UG=-SCREEN
D=SIN(ARCCOS(UG))
COSB=N/D
IF (COSB.GE.1.) COSB=SCREEN
IF (COSB.LE.-1.) COSB=-SCREEN
B=ARCCOS(COSB)
TEST=SITE(1,K)-LAMBDA
1000 IF (TEST) 1000,1300,1100
1100 IF (TEST+PI) 1200,1300,1300
1200 IF (TEST-PI) 1200,1300,1300
A=-1
C=TWOPi
1300 RETURN
END

```

```

SUBROUTINE SHIFT(FROML,FRMMP,TOL,TOP,NRTYPE,DEGF,DIST)
C THIS SUBROUTINE COMPUTES AND WRITES SEARCH PROGRESS
DIMENSION TYPES(2)
DATA RADDEG/57.29578/, FNCECC/1.006768/
DATA TYPES/'GOLD','GRAD'/
CALL AZDIST(FROML,FRMMP,TOL,TYPE,AZGO,DIST)
DEGLAM=TOL*RADDEG
DEGPHI=ATAN(FNCECC*TAN(TOP))*RADDEG
IF (NRTYPE.EQ.2) GO TO 1000
TYPE=TYPES(1)
GO TO 1100
1000 TYPE=TYPES(2)
1100 WRITE (6,1) TYPE,DIST,AZGO,DEGLAM,DEGPHI,DEGF
1 FORMAT (31X,A4,F7.1,' NM ON AZ',F6.1,' TO',F8.3,'E',
1 F8.3,'. F=F10.3)
FROML=TOL
FRMMP=TOP
RETURN
END

```

```

SUBROUTINE PLOT(LAMBDA,PSI,DEGF,NUMBER)
C PLCT COMPUTES AND WRITES THE CONFIDENCE REGION
REAL LAMBDA,L
DOUBLE PRECISION VARPSI,PIHALF,PI,ARC,COSCOL,COLAT,
1 CCSANG
COMMNC/SIGNAL/THETA(19)
DIMENSION OUT(52),SYMBOL(3)
DIMENSION CHI900(19)
DIMFNSION JSCALE(11)
DATA SYMBOL/*',',',S'
DATA PIHALF/1.570795326D0/,PI/3.141592653D0/
DATA DEGRAD/1.745329E-2/,TWOPi/6.283185/,
1 RADDEG/57.29578/
DATA CHI900/2.714.61,6.25,7.78,9.24,10.6,12.0,13.4,
14.7,16..17.3,18.5,19.8,21.1,22.3,23.5,24.8,26.,27.2/
BCUND=CHI900(NUMBER)
PROB=.900
NSIZE=400
0900 SIZE=NSIZE/2
TOP=PSI+1.05*SIZE*2.9070443E-4
KCUNT=0
MARKV=0
DO 2700 I = 1,41
ADD=0.
IF (MARV.NE.2) GO TO 1100
DO 1000 J = 1,51
OUT(J)=SYMBOL(1)
1000 CONTINUE
GO TO 2500
1100 TAKE=I
DIRECT=1.
VARPSI=TOP-TAKE*0.05*SIZE*2.9070443E-4
IF (VARPSI.LT.PIHALF) GO TO 1200
VARPSI=PI-VARPSI

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DIRECT=-1.
ADD=PI
GO TO 1300
1200 IF (VARPSI.GT.-PIHALF) GO TO 1300
VARPSI=-PI-VARPSI
DIRECT=-1.
ADD=PI
1300 PARC=1.04*SIZE*2.9070443E-4
MARKH=0
DO 2500 J = 1,51
IF (MARKH.EQ.2) GO TO 2100
TAKE=J
ARC=PARC-TAKE*0.04*SIZE*2.9070443E-4
COSCOL=DCOS(ARC)*DSIN(VARPSI)
IF (COSCOL.GT.1.D0) COSCOL=1.D0
IF (COSCOL.LT.-1.D0) COSCOL=-1.D0
COLAT=DARCCS(COSCOL)
P=PIHALF-COLAT
COSANG=DCOTAN(VARPSI)*DCOTAN(COLAT)
IF (COSANG.GT.1.D0) COSANG=1.D0
IF (COSANG.LT.-1.D0) COSANG=-1.D0
ANGLE=DARCCS(COSANG)
IF (J.LE.26) GO TO 1400
L=LAMBDA+DIRECT*ANGLE+ADD
GO TO 1500
1400 L=LAMBDA-DIRECT*ANGLE+ADD
CONTINUE
1500 IF (L.GT.TWOPi) L=L-TWOPi
IF (L.LT.0.) L=TWOPi+L
COSLAM=COS(L)
SINLAM=SIN(L)
COSPSI=COS(P)
SINPSI=SIN(P)
DEGF=0.
DO 2000 K = 1,NUMBER
CALL AZCCMP(COSLAM,SINLAM,COSPSI,SINPSI,K,L,A,B,C)
TEST=A*B+C-THETA(K)
IF (ABS(TEST)-PI) 1900,1900,1600
1600 IF (TEST) 1700,1900,1800
1700 C=C+TWOPi
GO TO 1900
1800 C=C-TWOPi
1900 OMEGA=A*B+C-THETA(K)
2000 CFGF=DEGF+(CMEGA*RADDEG)**2
2100 OUT(J)=SYMBOL(1)
MARKER=MARKH+1
GO TO (2300,2200,2500), MARKER
2200 IF (DEGF.GT.BOUND) MARKH=2
2300 IF (DEGF.GT.BOUND) GO TO 2400
OUT(J)=SYMBOL(2)
KOUNT=KOUNT+1
MARKH=1
MARKV=1
IF (I.GT.1) GO TO 2350
NSIZE=NSIZE+200
GO TO 0900
2350 IF (J.GT.1) GO TO 2400
NSIZE=NSIZE+NSIZE
WRITE (6,7)
7 FORMAT ('1')
GO TO 0900
2400 IF (I.NE.21) GO TO 2500
IF (J.EQ.26) OUT(J)=SYMBOL(3)
2500 CONTINUE
IF (MARKV.NE.1) GO TO 2600
IF (MARKH.EQ.0) MARKV=2
2600 CONTINUE
IF (I.NE.1) GO TO 2650
WRITE (6,1) PROB,NSIZE,NSIZE
1 FORMAT ('0',30X,F6.3,' CONFIDENCE REGION PLOT ON',I5,
1' NM BY',I5,' NM AREA')
WRITE (6,3)

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```
3      FORMAT (37X,17('---'),'---')
2650    CONTINUE
        VERT=21-I
        NM=ABS(0.05*VERT*SIZE)+.5
        IF (I.GT.21) NM=-NM
        WRITE (6,4) NM,(CUT(J),J=1,51)
4      2700    FORMAT (' ',30X,I5,' I',51A1,' I')
        CONTINUE
        WRITE (6,3)
        DO 2800 J = 1,11
        ADD=J
        JSSCALE(J)=ABS((0.2*ADD-1.2)*SIZE)+.5
2800    IF (J.LT.6) JSSCALE(J)=-JSSCALE(J)
        WRITE (6,5) (JSCALE(J),J=1,11)
5      FORMAT (34X,11I5)
        IJAREA=0.002*SIZE**2+.5
        NMAREA=KCUNT*IJAREA
        WRITE (6,6) NMAREA
6      FORMAT (30X,' CONFIDENCE REGION AREA =',I7,
1' NAUTICAL MILES SQUARED.')
        RETURN
        END
```

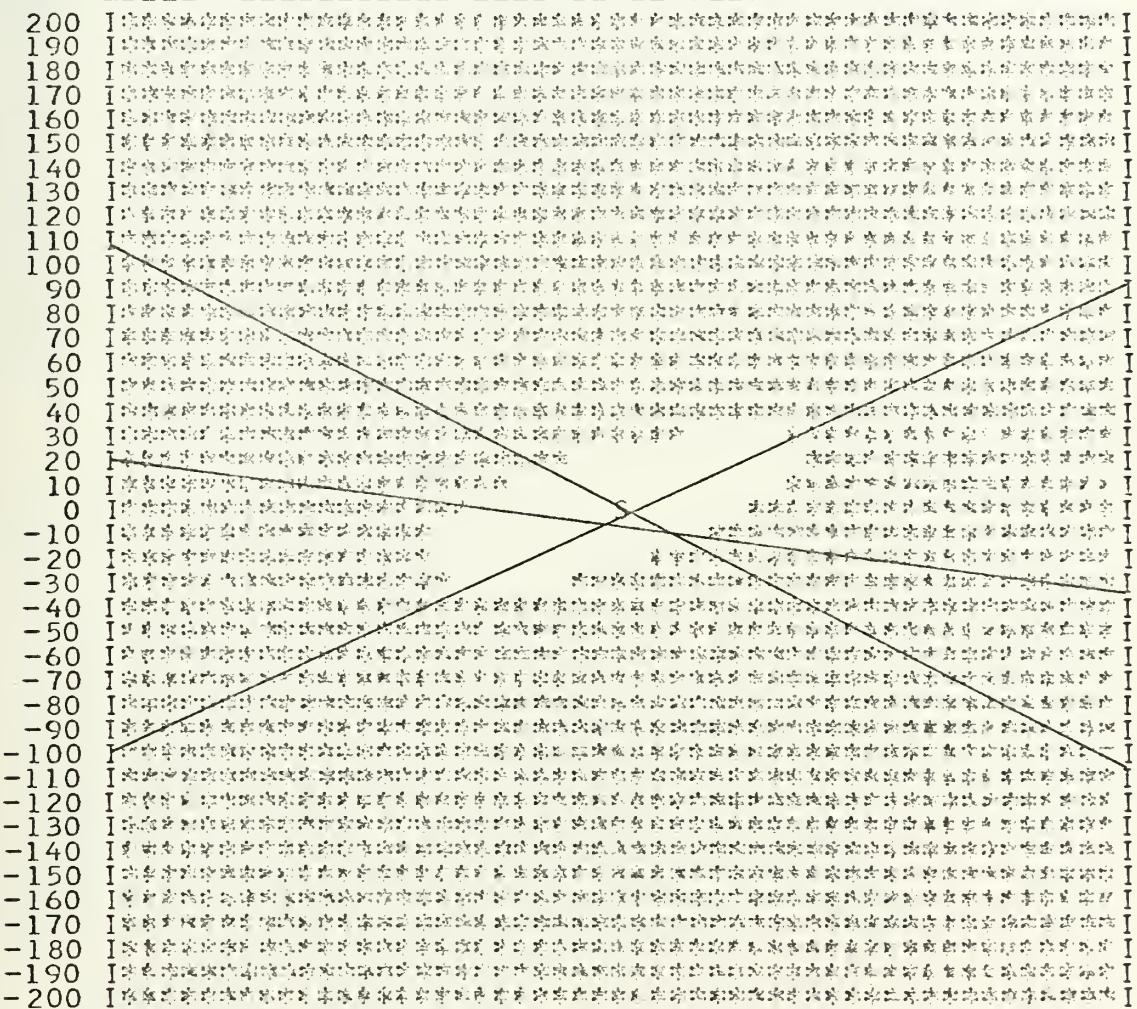

GOLD	4398.4	NM	ON	AZ	270.0	TO	286.740E	0.0	.	F=	20254.277
GOLD	4675.4	NM	ON	AZ	180.0	TO	286.740E	-77.953.	F=	11288.250	
GOLD	1021.9	NM	ON	AZ	134.1	TO	16.323E	-77.953.	F=	8562.371	
GOLD	1898.3	NM	ON	AZ	0.0	TO	16.323E	-46.449.	F=	3712.686	
GOLD	1108.5	NM	ON	AZ	99.8	TO	43.155E	-46.449.	F=	868.700	
GOLD	412.8	NM	ON	AZ	0.0	TO	43.155E	-39.571.	F=	111.312	
GRAD	271.3	NM	ON	AZ	94.0	TO	49.004E	-39.741.	F=	15.526	
GRAD	84.6	NM	ON	AZ	53.8	TO	50.461E	-38.898.	F=	0.158	
GRAD	10.2	NM	ON	AZ	76.8	TO	50.673E	-38.859.	F=	0.050	
GRAD	0.1	NM	ON	AZ	52.7	TO	50.675E	-38.858.	F=	0.050	

50-40E, 38-51S IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 0.05 DEGREES SQUARED.
 3 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	AZIMUTH	AZIMUTH	NM	NORM	NM	AZ
9	HONOLULU	BETA	- THETA = OMEGA	228.098	228.241	-0.142	9001 27.1 4 117.1
19	WICHITA	102.627	102.797	-0.170	9301	187.3	4 277.3
10	LOS ANGELES	121.457	121.424	0.033	10199	335.1	0 245.1

MEAN ARC STATISTIC = 21.8 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 34.7 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 400 NM BY 400 NM AREA



-200 -160 -120 -80 -40 0 40 80 120 160 200
 CONFIDENCE REGION AREA = 6080 NAUTICAL MILES SQUARED.

GOLD	6104.5	NM	ON	AZ	270.0	TO	258.323E	0.0	F=	86.016
GOLD	276.9	NM	ON	AZ	180.0	TO	258.323E	-4.643	F=	79.381
GRAD	2293.2	NM	ON	AZ	167.1	TO	268.963E	-41.796	F=	10.165
GRAD	1199.7	NM	ON	AZ	160.2	TO	282.300E	-59.978	F=	0.043
GRAD	77.3	NM	ON	AZ	151.7	TO	283.557E	-61.102	F=	0.013
GRAD	0.9	NM	ON	AZ	97.3	TO	283.589E	-61.104	F=	0.013
GRAD	0.0	NM	ON	AZ	0.0	TO	283.590E	-61.104	F=	0.013

76-25W, 61- 6S IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 0.01 DEGREES SQUARED.
 3 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	AZIMUTH	AZIMUTH	RANGE	NM	NORM	NM	AZ
7	EL PASO	165.841	165.922	BETA - THETA = OMEGA	5747	244.6	4	334.6
17	RENO	159.655	159.655	0.000	6383	56.4	0	326.4
8	HELENA	162.349	162.270	0.080	6673	64.5	4	334.5

MEAN ARC STATISTIC = 75.6 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 5.5 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 2400 NM BY 2400 NM AREA



-1200 -960 -720 -480 -240 0 240 480 720 960 1200
 CONFIDENCE REGION AREA = 311040 NAUTICAL MILES SQUARED.

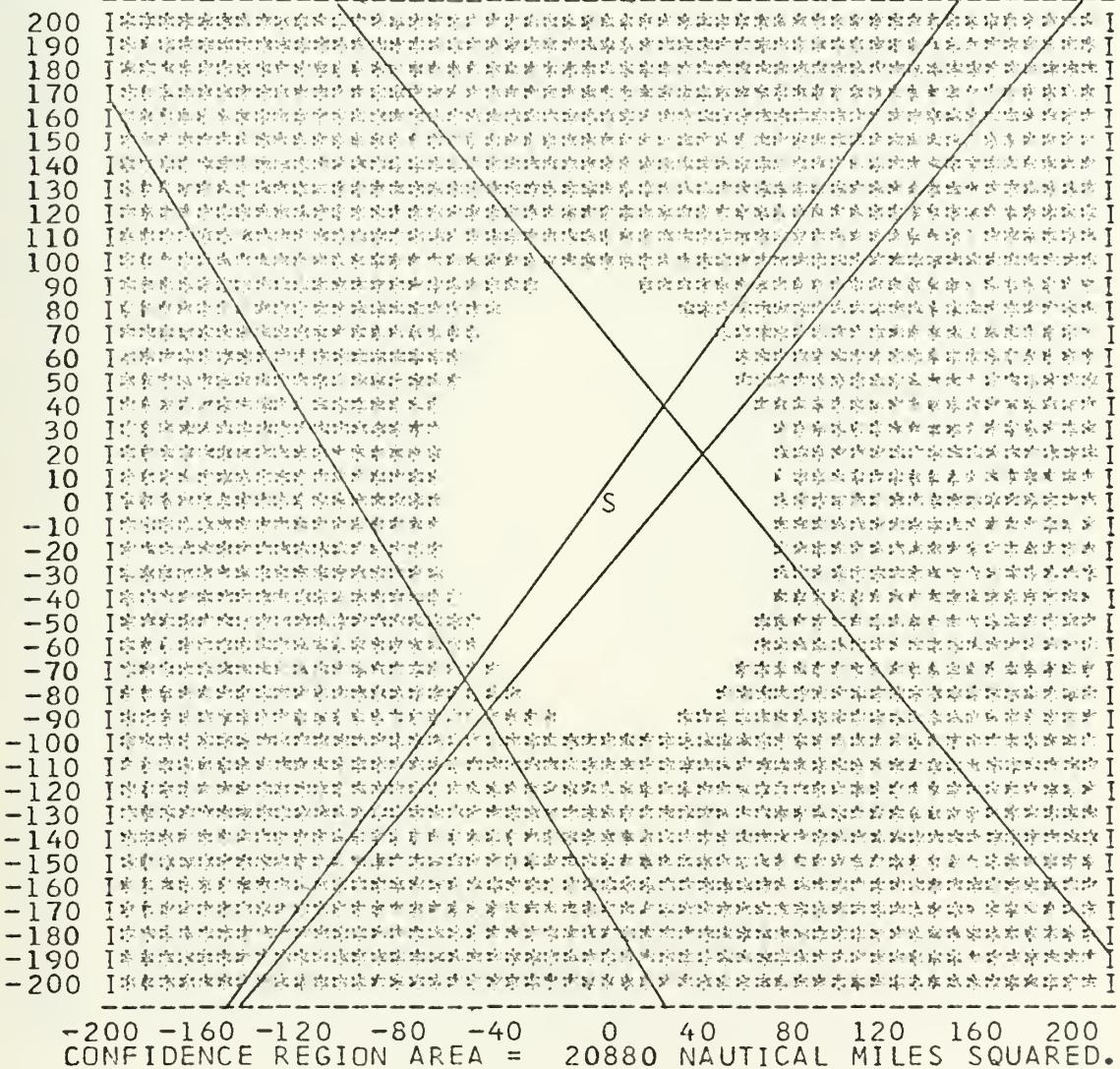
GOLD	7810.8	NM	ON	AZ	270.0	TO	229.901E	0.0	.	F=	663.993
GOLD	995.3	NM	ON	AZ	180.0	TO	229.901E	-16.684.	F=	13.884	
GOLD	97.0	NM	ON	AZ	90.2	TO	231.588E	-16.684.	F=	3.408	
GOLD	3.8	NM	ON	AZ	0.0	TO	231.588E	-16.620.	F=	3.381	
GRAD	10.1	NM	ON	AZ	67.7	TO	231.750E	-16.556.	F=	3.289	
GRAD	0.0	NM	ON	AZ	0.0	TO	231.750E	-16.556.	F=	3.289	

128-15W, 16-33S IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 3.29 DEGREES SQUARED.
 4 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	AZIMUTH	AZIMUTH		NM	NORM	NM	AZ
14	NASHVILLE	BETA -	THETA = OMEGA		3929	306.1	6	36.1
2	ATTU	224.264	224.377	-0.113	5135	238.9	86	328.9
9	HONOLULU	124.892	126.340	-1.448	2854	50.6	46	320.6
16	NORFOLK	140.026	138.983	1.043	4406	130.0	17	40.0
		232.077	231.774	0.303				

MEAN ARC STATISTIC = 68.0 DEGREES.
 MFAN ACUTE ANGLE OF INTERSECTION = 50.9 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 400 NM BY 400 NM AREA



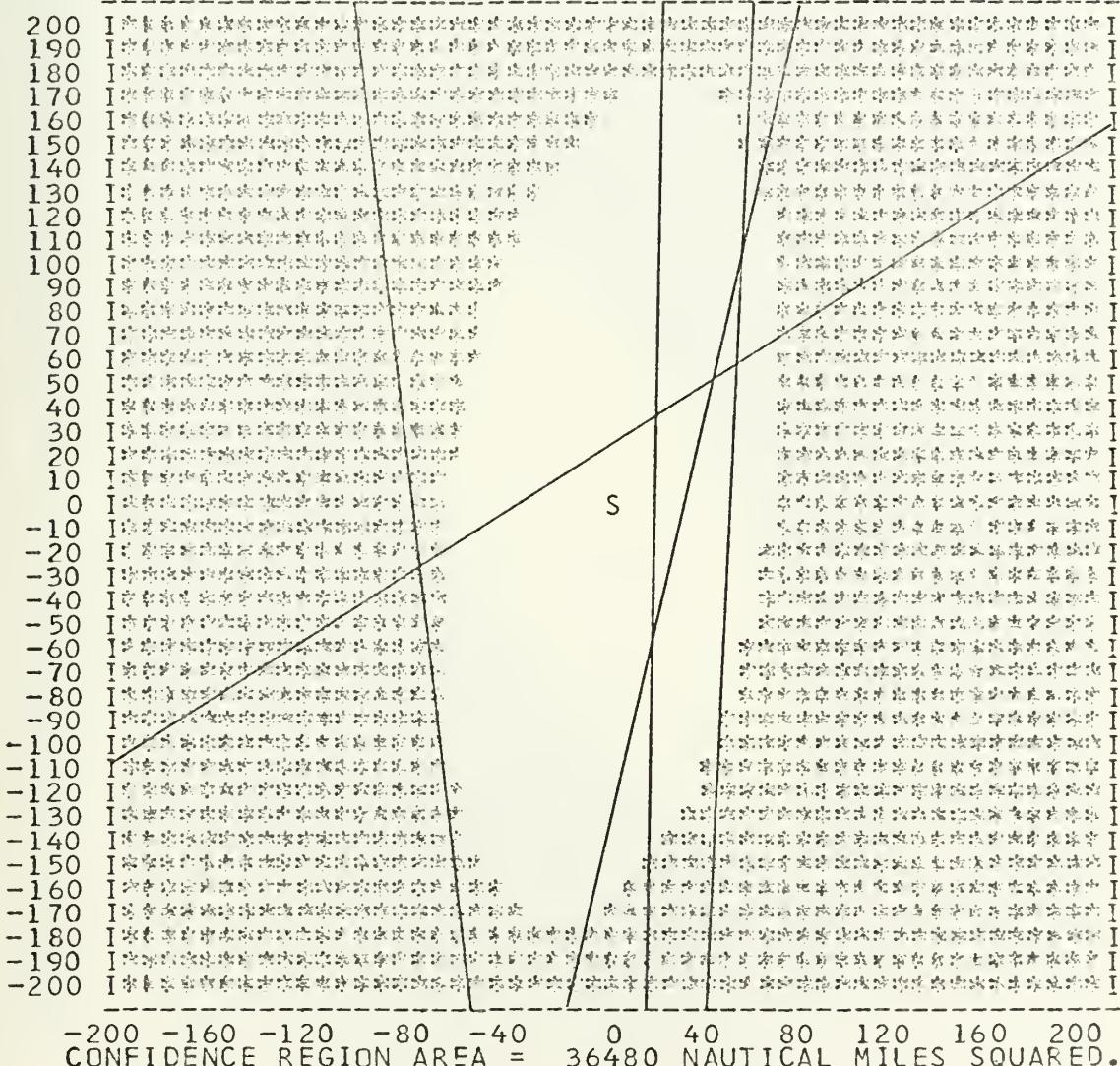
GOLD	4399.1	NM	ON	AZ	90.0	TO	73.272E	0.0	.	F=	767.351
GOLD	1765.6	NM	ON	AZ	0.0	TO	73.272E	29.574.	F=	26.527	
GOLD	123.9	NM	ON	AZ	89.4	TO	75.642E	29.574.	F=	4.538	
GOLD	71.1	NM	ON	AZ	0.0	TO	75.642E	30.763.	F=	3.487	
GRAD	14.2	NM	ON	AZ	55.7	TO	75.869E	30.897.	F=	3.303	
GRAD	0.2	NM	ON	AZ	40.6	TO	75.871E	30.899.	F=	3.303	
GRAD	0.0	NM	ON	AZ	0.0	TO	75.871E	30.899.	F=	3.303	

75-52E, 30-54N IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 3.30 DEGREES SQUARED.
 5 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	AZIMUTH	AZIMUTH		RANGE NM	NORM	NM	AZ
17	RENO	BETA -	THETA =	OMEGA	6511	102.9	27	12.9
5	DENVER	345.832	345.356	0.476	6589	90.8	16	0.8
12	MIDWAY	359.185	358.897	0.288	5329	327.7	22	57.7
19	WICHITA	304.587	304.956	-0.369	6693	263.7	81	353.7
7	EL PASO	6.275	7.726	-1.451	7064	92.7	46	2.7
		357.715	356.849	0.866				

MEAN ARC STATISTIC = 72.3 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 28.0 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 400 NM BY 400 NM AREA



-200 -160 -120 -80 -40 0 40 80 120 160 200
 CONFIDENCE REGION AREA = 36480 NAUTICAL MILES SQUARED.

GOLD	2476.5	NM	ON	AZ	90.0	TO	41.249E	0.0	.	F=	258.677
GOLD	561.3	NM	ON	AZ	0.0	TO	41.249E	9.411.	.	F=	94.282
GOLD	207.1	NM	ON	AZ	270.3	TO	37.754E	9.411.	.	F=	42.317
GOLD	213.3	NM	ON	AZ	0.0	TO	37.754E	12.985.	.	F=	20.378
GRAD	277.8	NM	ON	AZ	318.0	TO	34.526E	16.421.	.	F=	0.264
GRAD	4.4	NM	ON	AZ	325.8	TO	34.483E	16.482.	.	F=	0.259
GRAD	0.1	NM	ON	AZ	325.7	TO	34.482E	16.484.	.	F=	0.259
GRAD	0.0	NM	ON	AZ	0.0	TO	34.482E	16.484.	.	F=	0.259

34-29E, 16-29N IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 0.26 DEGREES SQUARED.
 5 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	AZIMUTH	AZIMUTH	RANGE	NM	NORM	NM	AZ
8	HELENA	34.383	33.967	BETA - THETA = OMEGA	6624	66.2	23	336.2
11	MIAMI	63.878	64.000	-0.122	6235	212.4	7	302.4
14	NASHVILLE	57.577	57.539	0.038	6229	44.6	2	314.6
17	RENO	28.527	28.765	-0.238	7165	247.2	12	337.2
1	ANCHORAGE	355.710	355.825	-0.114	6153	272.1	6	2.1

MEAN ARC STATISTIC = 72.1 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 28.4 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 600 NM BY 600 NM AREA



-300 -240 -180 -120 -60 0 60 120 180 240 300
 CONFIDENCE REGION AREA = 57600 NAUTICAL MILES SQUARED.

GOLD	1690.8	NM	ON	AZ	270.0	TO	331.837E	0.0	F=	102.897
GOLD	141.5	NM	ON	AZ	180.0	TO	331.837E	-2.373	F=	83.012
GRAD	629.1	NM	ON	AZ	135.7	TO	339.241E	-9.884	F=	3.863
GRAD	2.2	NM	ON	AZ	51.3	TO	339.270E	-9.861	F=	3.846
GRAD	0.1	NM	ON	AZ	48.4	TO	339.271E	-9.860	F=	3.845

20-44W, 9-52S IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 3.85 DEGREES SQUARED.
 6 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	AZIMUTH	AZIMUTH		NM	NORM	NM	AZ
	BETA	-	THETA = OMEGA					
3	BOSTON	125.857	125.554	0.303	4166	52.3	17	322.3
12	MIDWAY	54.831	54.455	0.376	9092	43.3	10	313.3
2	ATTU	19.885	19.173	0.712	8140	78.4	29	348.4
11	MIAMI	113.543	114.172	-0.629	4071	213.2	34	303.2
1	ANCHORAGE	58.837	60.428	-1.591	7009	244.4	85	334.4
10	LOS ANGELES	94.063	93.642	0.421	6103	32.9	24	302.9

MEAN ARC STATISTIC = 58.6 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 22.0 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 400 NM BY 400 NM AREA



-200 -160 -120 -80 -40 0 40 80 120 160 200
 CONFIDENCE REGION AREA = 26720 NAUTICAL MILES SQUARED.

GOLD	7158.7	NM	ON	AZ	90.0	TO	119.236E	0.0	.	F=	318.482
GOLD	251.0	NM	ON	AZ	0.0	TO	119.236E	4.209.	F=	261.669	
GRAD	1861.3	NM	ON	AZ	29.7	TO	136.486E	30.744.	F=	34.133	
GRAD	525.6	NM	ON	AZ	47.2	TO	144.451E	36.464.	F=	8.503	
GRAD	104.9	NM	ON	AZ	34.4	TO	145.701E	37.901.	F=	8.042	
GRAD	5.3	NM	ON	AZ	222.2	TO	145.627E	37.836.	F=	8.041	
GPAD	0.6	NM	ON	AZ	222.0	TO	145.618E	37.829.	F=	8.041	
GRAD	0.4	NM	ON	AZ	221.9	TO	145.612E	37.824.	F=	8.041	
GRAD	0.1	NM	ON	AZ	221.8	TO	145.611E	37.822.	F=	8.041	
GRAD	0.0	NM	ON	AZ	0.0	TO	145.611E	37.822.	F=	8.041	

145-37E, 37-49N IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 8.04 DEGREES SQUARED.
 6 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	BETA	-	AZIMUTH	NM	NORM	NM	AZ
16	NORFOLK	327.990	328.189	-0.199	5685	301.6	11	31.6
4	CORPUS CHRIS	315.081	314.619	0.462	5524	142.2	27	52.2
11	MIAMI	324.266	322.103	2.164	6215	132.2	126	42.2
18	SEATTLE	298.750	300.372	-1.622	3873	319.2	87	49.2
6	DETROIT	323.507	324.197	-0.690	5328	303.9	41	33.9
13	MINNEAPOLIS	316.823	316.840	-0.016	4923	307.8	0	37.8

MEAN ARC STATISTIC = 80.8 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 10.2 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 600 NM BY 600 NM AREA



-300 -240 -180 -120 -60 0 60 120 180 240 300
 CONFIDENCE REGION AREA = 37080 NAUTICAL MILES SQUARED.

GOLD	2341.5	NM	ON	AZ	90.0	TO	39.000E	0.0	.	F=	5868.273
GOLD	575.9	NM	ON	AZ	180.0	TO	39.000E	-9.656.	F=	5288.270	
GOLD	486.3	NM	ON	AZ	90.7	TO	47.215E	-9.656.	F=	4666.781	
GOLD	531.3	NM	ON	AZ	180.0	TO	47.215E	-18.558.	F=	3902.349	
GOLD	406.6	NM	ON	AZ	91.1	TO	54.355E	-18.558.	F=	3082.319	
GOLD	425.4	NM	ON	AZ	180.0	TO	54.355E	-25.695.	F=	2172.056	
GRAD	577.5	NM	ON	AZ	138.9	TO	61.849E	-32.774.	F=	215.962	
GRAD	95.2	NM	ON	AZ	36.1	TO	62.942E	-31.483.	F=	4.374	
GRAD	10.9	NM	ON	AZ	99.8	TO	63.151E	-31.514.	F=	2.152	
GRAD	0.1	NM	ON	AZ	25.4	TO	63.152E	-31.512.	F=	2.152	
GRAD	0.0	NM	ON	AZ	0.0	TO	63.152E	-31.512.	F=	2.152	

63- 9E, 31-31S IS THE LEAST SQUARES SOLUTION.

OBJECTIVE FUNCTION = 2.15 DEGREES SQUARED.

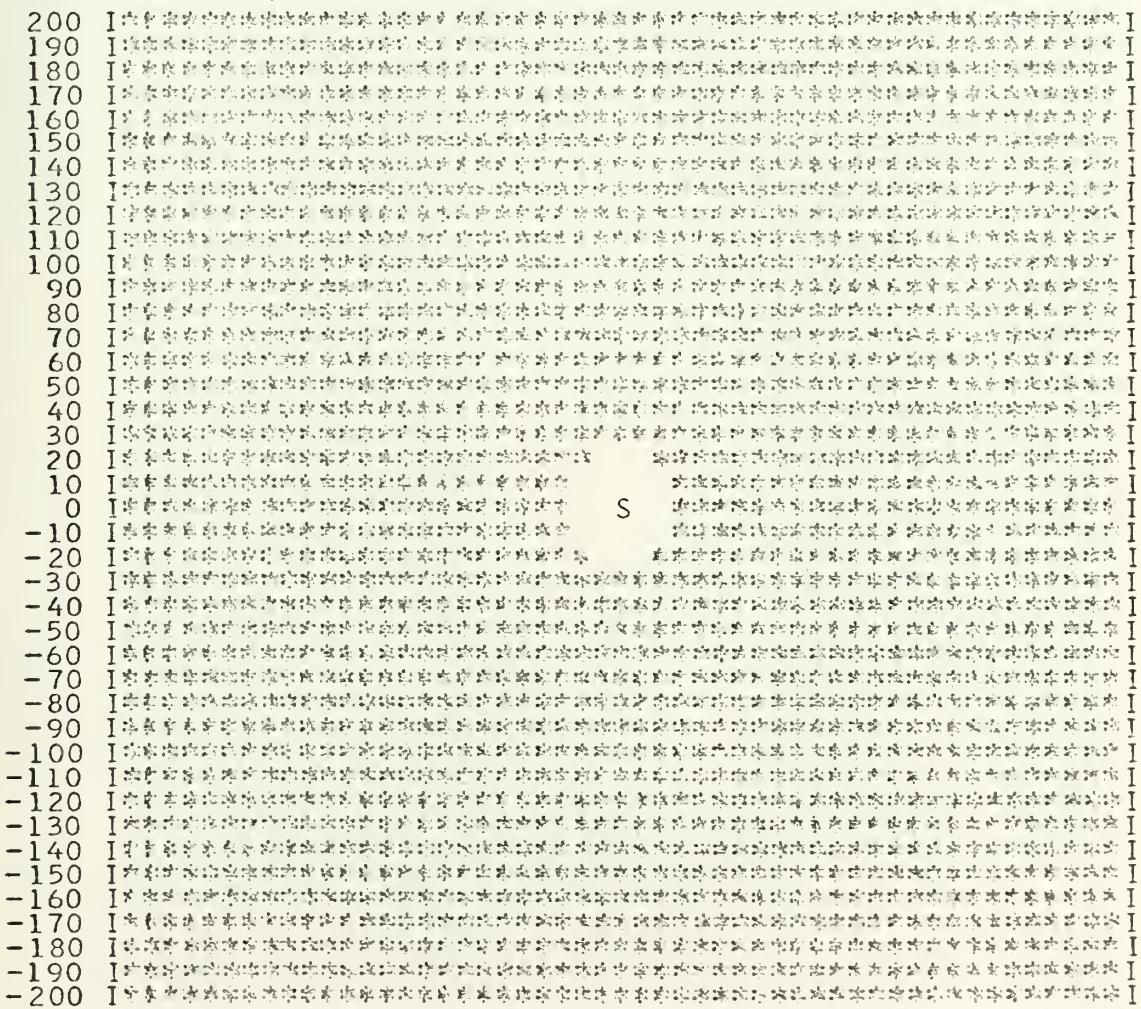
7 DEGREES OF FREEDOM

ID	INTERCEPT	SITE NAME	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
			AZIMUTH	AZIMUTH		NM	NORM	NM	AZ
			BETA	- THETA = OMEGA					
NR									
5	DENVER		53.228	53.230	-0.002	10049	223.7	0	313.7
15	NEW ORLEANS		100.649	101.286	-0.637	9425	182.5	14	272.5
6	DETROIT		79.209	78.237	0.972	9070	32.3	28	302.3
16	NORFOLK		90.386	89.567	0.820	8779	24.3	27	294.3
7	EL PASO		91.177	91.369	-0.192	10276	184.1	1	274.1
17	RENO		342.343	342.395	-0.051	10306	285.9	0	15.9
8	HELENA		15.592	15.892	-0.300	9875	257.2	4	347.2

MEAN ARC STATISTIC = 18.7 DEGREES.

MEAN ACUTE ANGLE OF INTERSECTION = 42.9 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 400 NM BY 400 NM AREA



-200 -160 -120 -80 -40 0 40 80 120 160 200
CONFIDENCE REGION AREA = 1680 NAUTICAL MILES SQUARED.

GOLD	5019.9	NM	ON AZ	270.0	TO	276.387E	0.0	F=	348.028
GOLD	819.4	NM	ON AZ	180.0	TO	276.387E	-13.737	F=	15.774
GOLD	73.1	NM	ON AZ	90.1	TO	277.641E	-13.737	F=	5.151
GOLD	71.1	NM	ON AZ	180.0	TO	277.641E	-14.929	F=	2.936
GRAD	1.4	NM	ON AZ	267.9	TO	277.616E	-14.930	F=	2.927
GRAD	0.1	NM	ON AZ	270.0	TO	277.615E	-14.930	F=	2.926

82-23W, 14-56S IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 2.93 DEGREES SQUARED.
 8 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	AZIMUTH	AZIMUTH		NM	NORM	NM	AZ
9	HONOLULU	BETA -	THETA = OMEGA		4943	204.5	36	294.5
17	RENC	109.222	109.844	-0.622	3868	58.4	25	328.4
6	DETROIT	139.373	138.901	0.472	3421	89.4	3	359.4
14	NASHVILLE	179.229	179.163	0.066	3060	84.8	43	354.8
3	BOSTON	174.501	173.573	0.928	3479	280.2	27	10.2
11	MIAMI	192.934	193.473	-0.539	2432	273.6	30	3.6
19	WICHITA	183.252	184.030	-0.778	3253	75.0	29	345.0
8	HELENA	162.091	161.478	0.613	4010	248.4	23	338.4
		148.667	149.088	-0.421				

MEAN ARC STATISTIC = 59.3 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 27.8 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 400 NM BY 400 NM AREA



-200 -160 -120 -80 -40 0 40 80 120 160 200
 CONFIDENCE REGION AREA = 33040 NAUTICAL MILES SQUARED.

GOLD	9485.0	NM	ON AZ	90.0	TO	157.984E	0.0	.	F=	215.253
GOLD	74.4	NM	ON AZ	0.0	TO	157.984E	1.248.	F=	207.475	
GRAD	1077.6	NM	ON AZ	51.3	TO	172.232E	12.394.	F=	6.162	
GRAD	13.2	NM	ON AZ	15.4	TO	172.292E	12.607.	F=	5.864	
GRAD	0.0	NM	ON AZ	0.0	TO	172.292E	12.607.	F=	5.864	

172-18E, 12-36N IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 5.86 DEGREES SQUARED.
 9 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	BETA	AZIMUTH	AZIMUTH	NM	NORM	NM	AZ
9	HONOLULU	257.935	258.530	-0.594	1788	339.6	17	69.6
10	LOS ANGELES	269.320	268.506	0.814	3968	147.9	44	57.9
11	MIAMI	289.107	288.596	0.511	5995	150.9	30	60.9
12	MIDWAY	213.947	213.857	0.090	1098	120.2	1	30.2
13	MINNEAPOLIS	282.050	282.304	-0.254	5064	315.3	15	45.3
14	NASHVILLE	286.504	286.771	-0.267	5480	322.6	16	52.6
15	NEW ORLEANS	284.612	282.885	1.727	5418	149.3	103	59.3
16	NORFOLK	293.348	293.874	-0.526	5903	317.3	31	47.3
17	RENO	265.894	266.982	-1.088	3907	322.7	59	52.7

MEAN ARC STATISTIC = 67.1 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 13.0 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 400 NM BY 400 NM AREA



-200 -160 -120 -80 -40 0 40 80 120 160 200
 CONFIDENCE REGION AREA = 18080 NAUTICAL MILES SQUARED.

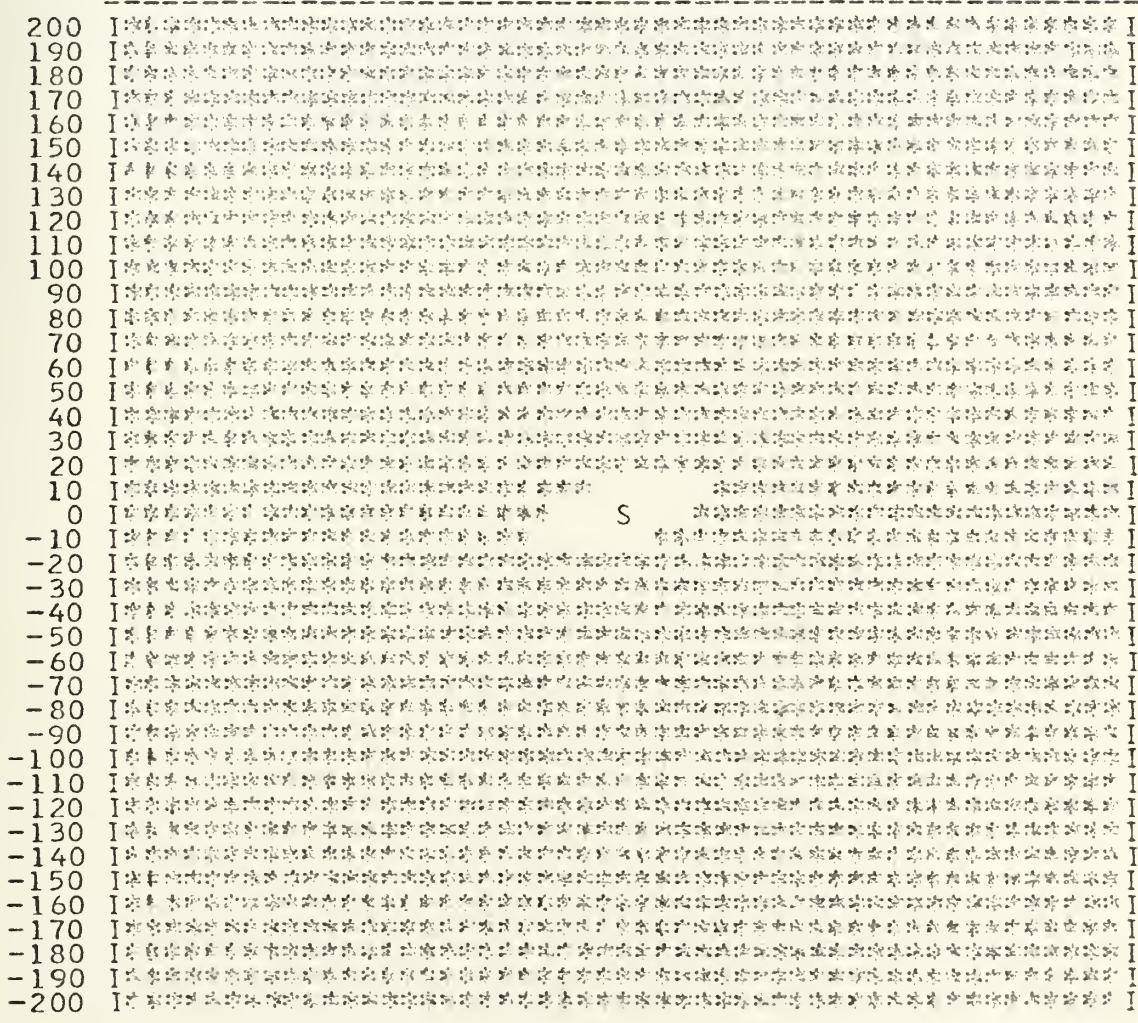
GOLD	202.5	NM	ON	AZ	90.0	TO	3.373E	0.0	.	F=	10268.355
GOLD	1955.9	NM	ON	AZ	0.0	TO	3.373E	32.769.	F=	5296.391	
GOLD	1314.5	NM	ON	AZ	277.1	TO	337.319E	32.769.	F=	3449.257	
GOLD	585.4	NM	ON	AZ	0.0	TO	337.319E	42.536.	F=	2502.434	
GOLD	2030.4	NM	ON	AZ	286.1	TO	290.972E	42.536.	F=	119.602	
GOLD	56.1	NM	ON	AZ	0.0	TO	290.972E	43.471.	F=	51.410	
GRAD	107.2	NM	ON	AZ	69.9	TO	293.299E	44.062.	F=	10.708	
GRAD	4.2	NM	ON	AZ	216.4	TO	293.241E	44.006.	F=	10.553	
GRAD	0.1	NM	ON	AZ	215.9	TO	293.240E	44.005.	F=	10.553	

66-46W, 44- ON IS THE LEAST SQUARES SOLUTION.
 OBJECTIVE FUNCTION = 10.55 DEGREES SQUARED.
 9 DEGREES OF FREEDOM

ID	INTERCEPT	SOLUTION	SIGNAL	SITE	RANGE	PLOT	THETA	VALS
NR	SITE NAME	AZIMUTH	AZIMUTH	OMEGA	NM	NORM	NM	AZ
	BETA	-	THETA					
6	DETROIT	76.495	77.395	-0.900	721	178.5	11	268.5
12	MIDWAY	42.761	41.147	1.614	5054	33.7	96	303.7
18	SEATTLE	74.220	72.903	1.317	2294	24.5	48	294.5
5	DENVER	68.794	67.963	0.832	1719	4.1	23	274.1
11	MIAMI	27.603	26.907	0.696	1274	304.7	15	214.7
17	RENO	65.683	66.781	-1.098	2357	193.2	41	283.2
4	CORPUS CHRIS	48.511	47.035	1.476	1765	335.6	43	245.6
10	LOS ANGELES	60.122	60.990	-0.868	2440	184.6	33	274.6
16	NORFOLK	50.167	50.470	-0.303	534	146.7	2	236.7

MEAN ARC STATISTIC = 33.6 DEGREES.
 MEAN ACUTE ANGLE OF INTERSECTION = 33.9 DEGREES.

0.900 CONFIDENCE REGION PLOT ON 400 NM BY 400 NM AREA



-200 -160 -120 -80 -40 0 40 80 120 160 200
 CONFIDENCE REGION AREA = 1520 NAUTICAL MILES SQUARED.

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WITH CONFIDENCE REGION AND LOCATION ERROR ANALYSIS

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13. ABSTRACT

Although parts of this paper may be applicable to various other location finding systems, radio direction finding was selected for development because of the extensive literature available on the subject. Using a modification of the navigation coordinate system, equations in spherical trigonometry are developed by analogy with graphical solution procedures. Ellipticity of the earth is corrected for by use of geocentric latitude in all calculations. The least squares solution is found by a combination of second-order gradient searches and Golden Section Searches on the corrected sphere. A Chi-Square contour confidence region is presented with defense against the empty region arguments and with verification of its reliability of target containment. Sensitivity of location error to network layout, number of measurements available and intercept geometry factors is examined by computer simulation, demonstrating the predictive value of these statistics. It is shown that the correlation between location error and the size of a reliable confidence region is too minimal to have predictive value.

KEY WORDS	LINK A		LINK B		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT
CHI-SQUARE CONTOUR CONFIDENCE REGION CONFIDENCE REGION DIRECTION FINDING NETWORKS GEOCENTRIC LATITUDE GOLDEN SECTION SEARCH HIGH FREQUENCY DIRECTION FINDING INTERCEPT GEOMETRY LEAST SQUARES FIX SOLUTION LOCATION ACCURACY PREDICTION LOCATION ERROR ANALYSIS LINES OF POSITION POSITION LOCATING RADIO DIRECTION FINDING SEARCH REGION SECOND-ORDER GRADIENT SEARCH SIMULATION						

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